

MATH 5453
Foundation of Fluid Dynamics

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Monday 12 noon Emmanuel Center SR7

Tuesday 12 noon " " "

Wednesday 2 pm EC Tower SR(8.60) — red route - staircase 4 → down 2 floors room behind

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|-------------------------|-----|---------------------|
| • Homework theory | 20% | } 3 Homework sheets |
| • Homework numerics | 20% | |
| • Homework experimental | 20% | |
| • Exam | 40% | |

Chapter I: Continuum hypothesis and kinematics

①

Fluid: substance that flows and that conforms to the shape of its container.

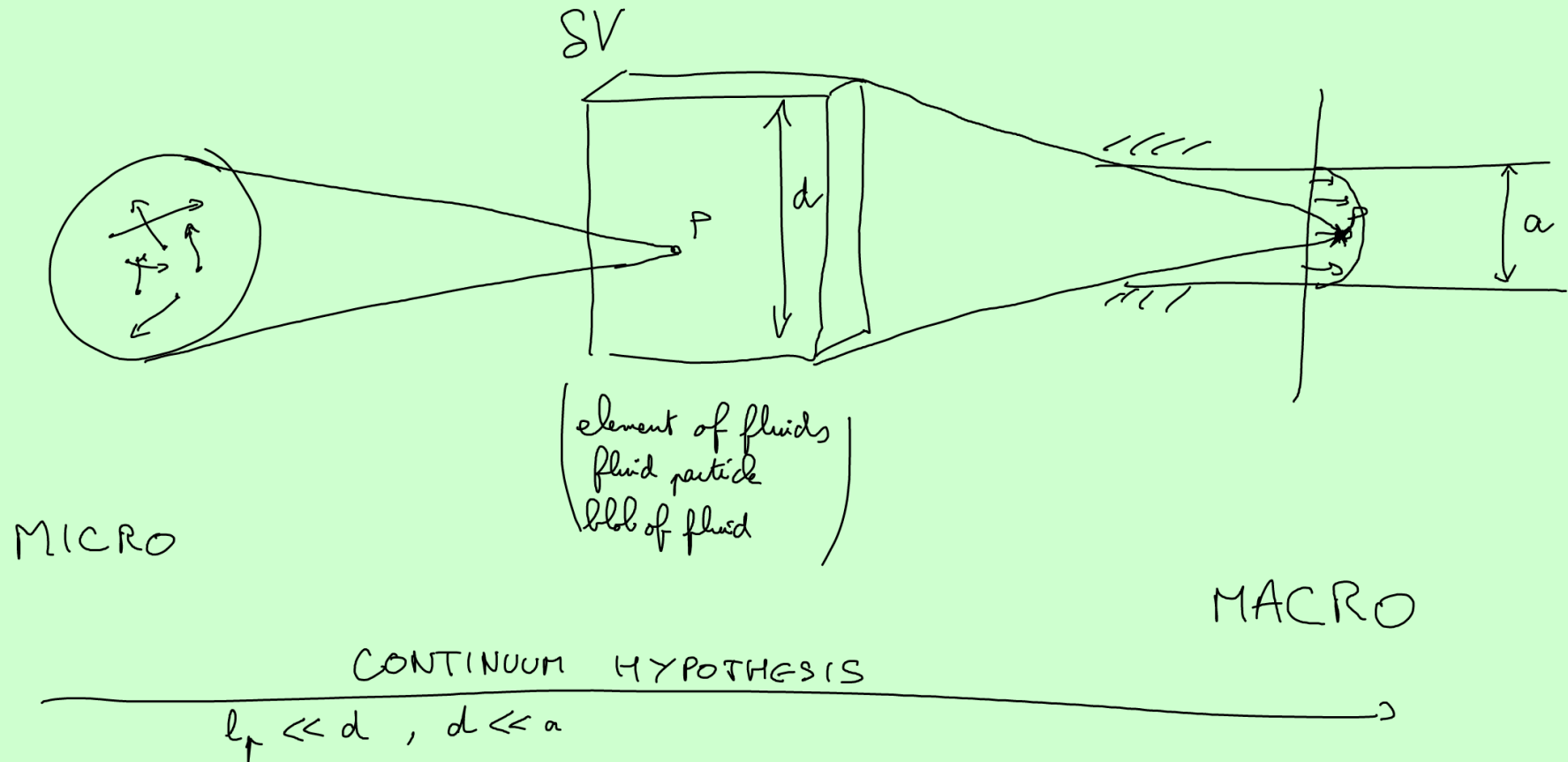
Dynamics: study of forces and torques when there is motion

Continuum hypothesis

1 cm³ of water contains 10^{23} particles

size of the particle: $1 \text{ \AA} = 10^{-10} \text{ m}$

mean free path $l_f = 10^{-8} \text{ m}$



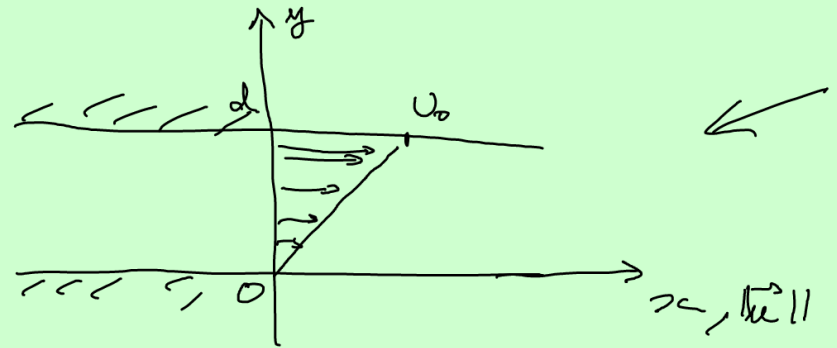
$$\vec{u}(\vec{P}, t) = \frac{\sum_{\text{particle in } \delta V} \vec{u}_{\text{particle}}}{\sum_{\text{particle in } \delta V} 1}$$

$$\rho(\vec{P}, t) = \frac{\sum_{\text{particle in } \delta V} m_{\text{particle}}}{\delta V}$$

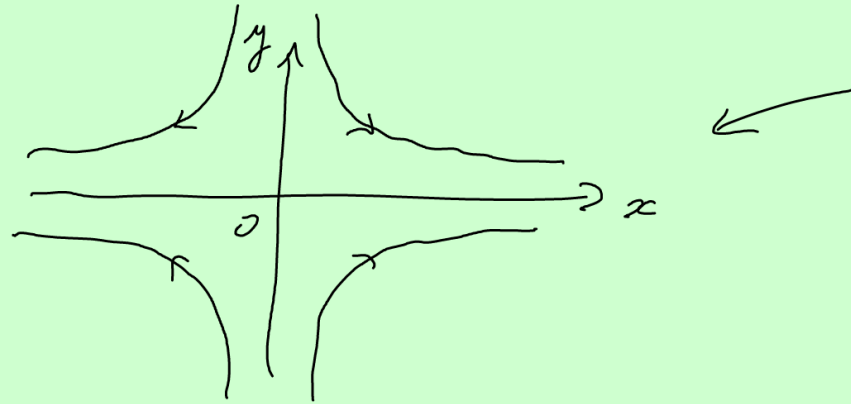
density

Simple flows

• Plane Couette flow: $\vec{u} = \frac{U_0 y}{d} \vec{e}_x$



• Stagnation flow: $\vec{u} = E x \vec{e}_x - E y \vec{e}_y$, $E > 0$



Time derivatives

• $\frac{\partial f}{\partial t}(\vec{x}, t)$: rate of change of f with time at $\vec{x}$

EULER

• $\frac{df}{dt}(\vec{x}(t), t) = \frac{\partial f}{\partial t} + \frac{\partial f}{\partial x} \frac{\partial x}{\partial t} + \frac{\partial f}{\partial y} \frac{\partial y}{\partial t} + \frac{\partial f}{\partial z} \frac{\partial z}{\partial t}$

$$= \frac{\partial f}{\partial t} + u \frac{\partial f}{\partial x} + v \frac{\partial f}{\partial y} + w \frac{\partial f}{\partial z}$$

$$= \frac{\partial f}{\partial t} + \underbrace{(\vec{u} \cdot \vec{\nabla}) f}_{\text{transport}} : \text{rate of change of } f \text{ with time as we follow a particle passing through } \vec{x} \text{ at time } t. \text{ LAGRANGE}$$

Particle paths

Visualization of the trajectory of a single particle (passive tracer),

Release a particle at $t=0$ and $\vec{x} = \vec{x}_0$

$$\frac{d\vec{x}}{dt} = \vec{u}(\vec{x}, t)$$

Example: Stagnation flow $\vec{u} = E x \vec{e}_x - E y \vec{e}_y$

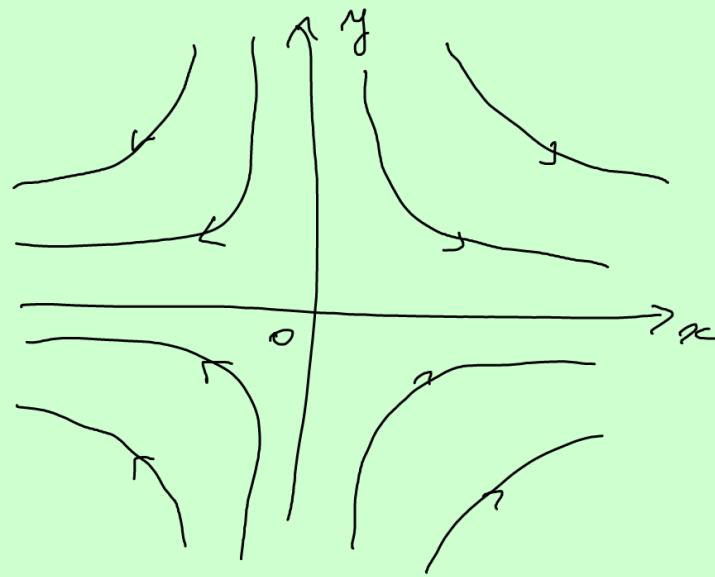
$$\Rightarrow \begin{cases} \frac{dx}{dt} = E x \\ \frac{dy}{dt} = -E y \end{cases}$$

$$\Rightarrow \begin{cases} x = x_0 e^{Et} \\ y = y_0 e^{-Et} \end{cases}$$

$$\Rightarrow xy = x_0 y_0$$

(2)

$$\Rightarrow y = \frac{x_0 y_0}{x}$$



Streamline : line everywhere tangent to the velocity

$$d\vec{c} = \vec{u} ds$$

$\swarrow$
 element of displacement
 along a streamline

 $\swarrow$
 arclength along the
 streamline

$$\Rightarrow \frac{d\vec{c}}{ds} = \vec{u}$$

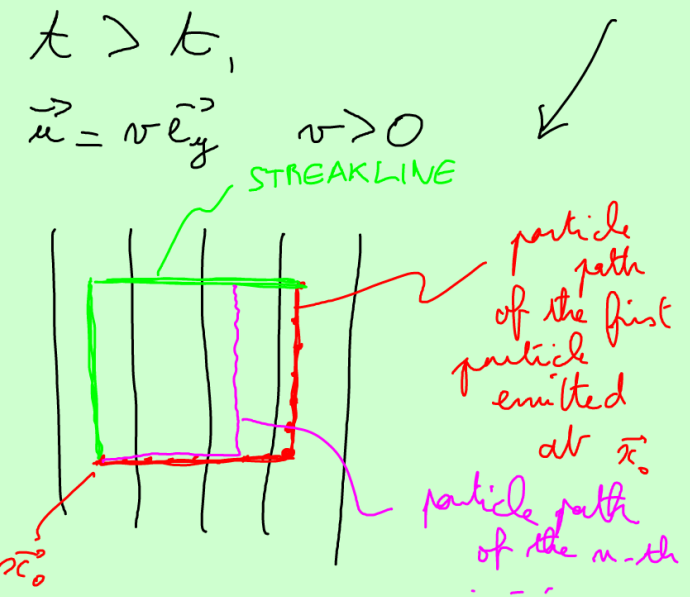
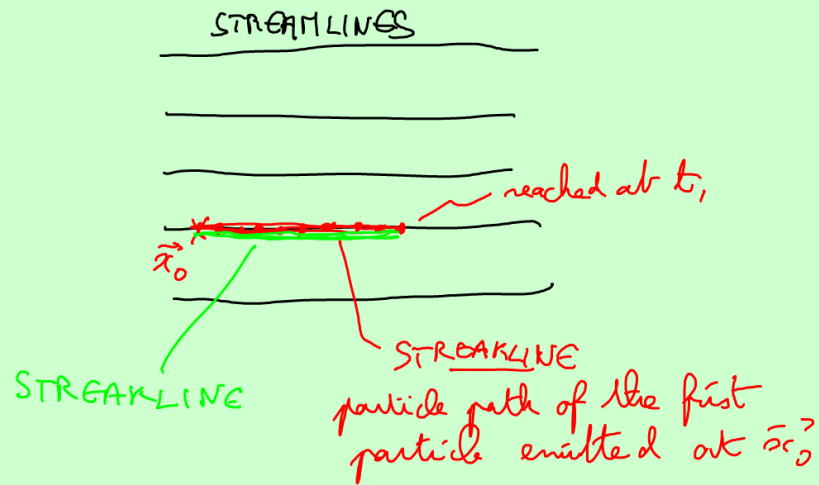
$$\Rightarrow \frac{dx}{u} = \frac{dy}{v} = \frac{dz}{w} = ds$$

If the flow is steady, streamlines and particle paths coincide.

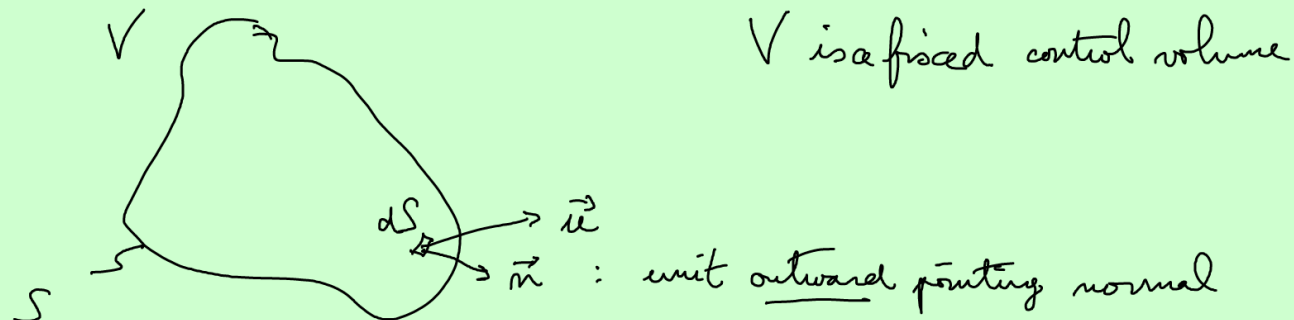
Streakline: Continuous release of dye from $\vec{x} = \vec{x}_0$ from $t = t_0$ to $t = T$. The line thereby drawn is a streakline.

$$t < t_1$$

$$\vec{u} = u \vec{e}_x \quad u > 0$$



Chapter 2: Mass conservation



total mass in V : $M_V = \int_V \rho \, dV$

Rate of change of mass in V : $\frac{dM_V}{dt} = \frac{d}{dt} \int_V \rho \, dV$

$$= \int_V \frac{\partial \rho}{\partial t} \, dV$$

M_V may change with time due to the flow of fluid through S :

$$\frac{dM_V}{dt} = - \int_S \rho \, \vec{u} \cdot \vec{n} \, dS$$

$$\Rightarrow \int_V \frac{\partial \rho}{\partial t} \, dV = - \int_S \rho \, \vec{u} \cdot \vec{n} \, dS$$

divergence theorem

$$\Rightarrow \int_V \frac{\partial \rho}{\partial t} \, dV = - \int_V \vec{\nabla} \cdot (\rho \vec{u}) \, dV$$

$$\Rightarrow \int_V \left[\frac{\partial \rho}{\partial t} + \vec{\nabla} \cdot (\rho \vec{u}) \right] dV = 0$$

We can take V to be the "blob" of fluid, so that the integrand has to vanish pointwise.

$$\Rightarrow \left\{ \begin{array}{l} \frac{d\rho}{dt} + \vec{\nabla} \cdot (\rho \vec{u}) = 0 \\ \frac{d\rho}{dt} + \rho \underbrace{\vec{\nabla} \cdot \vec{u} + (\vec{u} \cdot \vec{\nabla})}_{\text{continuity equation mass conservation}} \rho = 0 \\ \frac{D\rho}{Dt} + \rho \vec{\nabla} \cdot \vec{u} = 0 \end{array} \right.$$

$\vec{B}$ $\vec{A}$
 $\underline{u}$

Incompressible fluid: a fluid which has constant density

$$\frac{D\rho}{Dt} = 0 \Rightarrow \rho \vec{\nabla} \cdot \vec{u} = 0$$

$$\Rightarrow \boxed{\vec{\nabla} \cdot \vec{u} = 0} \quad \text{incompressibility} \left\{ \begin{array}{l} \text{condition} \\ \text{constraint} \end{array} \right.$$

We know that $\vec{\nabla} \cdot (\underbrace{\vec{\nabla} \times \vec{\varphi}}_{\vec{u}}) = 0$

$$\Rightarrow \vec{u} = \vec{\nabla} \times \vec{\varphi} \quad \text{streamfunction}$$

Examples in two dimensions: $\vec{u} = u(x, y) \vec{e}_x + v(x, y) \vec{e}_y$

$$\hookrightarrow \vec{\varphi} = \varphi(x, y) \vec{e}_z$$

$$u = \frac{\partial \varphi}{\partial y} \quad v = -\frac{\partial \varphi}{\partial x}$$

$$\vec{\nabla} \times \vec{\varphi} = \begin{pmatrix} \frac{\partial}{\partial x} \\ \frac{\partial}{\partial y} \end{pmatrix} \times \begin{pmatrix} 0 \\ 0 \\ \varphi \end{pmatrix}$$

(3)

streamline: line where ψ is constant

$$\bullet d\psi = 0 \Rightarrow \frac{\partial \psi}{\partial x} dx + \frac{\partial \psi}{\partial y} dy = 0$$

$$\Rightarrow -v dx + u dy = 0$$

$$\Rightarrow \vec{u} \times d\vec{l} = \vec{0} \quad d\vec{l} = dx \vec{e}_x + dy \vec{e}_y$$

element of displacement along the streamline

$$\Rightarrow \vec{u} \parallel d\vec{l} \text{ (streamline)}$$

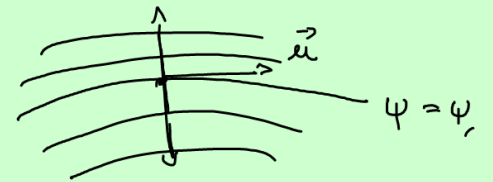
$$\bullet \vec{u} \cdot \vec{\nabla} \psi = u \frac{\partial \psi}{\partial x} + v \frac{\partial \psi}{\partial y}$$

$$= -uv + vu$$

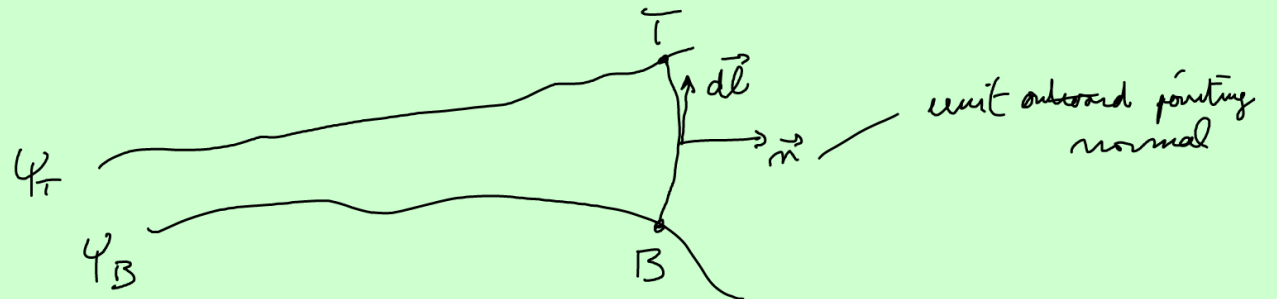
$$= 0$$

$$\Rightarrow \vec{u} \perp \text{direction of fastest increase of } \psi$$

$$\Rightarrow \vec{u} \text{ tangent to the streamline}$$



flux between streamlines:



$$Q = \int_B^T \vec{u} \cdot \underbrace{\vec{n}}_{\text{normal}} ds$$



$$d\vec{l} = dx \vec{e}_x + dy \vec{e}_y$$

element of displacement
along the (arbitrary)

$$\begin{aligned} \vec{n} ds &= dy \vec{e}_x - dx \vec{e}_y \\ &= ds \left(\frac{dy}{ds} \vec{e}_x - \frac{dx}{ds} \vec{e}_y \right) \end{aligned}$$

TB line

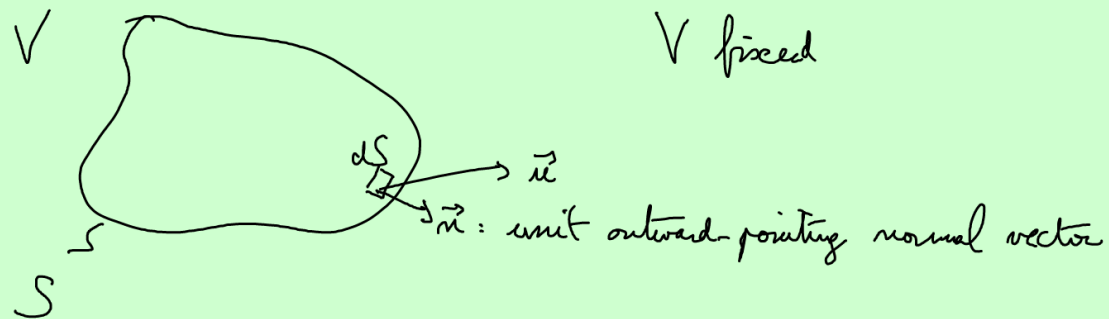
$$Q = \int_B^T \left[u \frac{dy}{ds} - v \frac{dx}{ds} \right] ds$$

$$= \int_B^T \left[\frac{\partial \psi}{\partial y} \frac{dy}{ds} + \frac{\partial \psi}{\partial x} \frac{dx}{ds} \right] ds$$

$$= \int_{\psi_B}^{\psi_T} d\psi$$

$$= \psi_T - \psi_B$$

Chapter 3; The Navier-Stokes equation



momentum density: $\vec{q} = \rho \vec{u}$

Total momentum in V : $\int_V \rho \vec{u} dV$

Rate of change of the total momentum in V with time: $\frac{d}{dt} \int_V \rho \vec{u} dV = \int_V \frac{\partial(\rho \vec{u})}{\partial t} dV$

$$\left(\begin{array}{c} \text{rate of change of} \\ \text{momentum in } V \end{array} \right) = \left(\begin{array}{c} \text{flow of momentum} \\ \text{through } S \end{array} \right) + \left(\begin{array}{c} \text{net force acting} \\ \text{on } V \end{array} \right)$$

$\uparrow$
Newton's 2nd law

I

II

III

$$\vec{I} = \int_V \frac{\partial}{\partial t} (\rho \vec{u}) dV \quad \longrightarrow \quad I_i = \int_V \frac{\partial}{\partial t} (\rho u_i) dV$$

$$\vec{\Pi} = - \int_S \vec{q} \vec{u} \cdot \vec{n} dS \quad \longrightarrow \quad \Pi_i = - \int_S q_i u_j n_j dS$$

$$- \int_S \rho \vec{u} \vec{u} \cdot \vec{n} dS$$

$$= - \int_V \vec{\nabla} \cdot (\rho \vec{u} \vec{u}) dV$$

$$= - \int_S \rho u_i u_j n_j dS$$

↓ divergence theorem

$$= - \int_V \frac{\partial}{\partial x_j} (\rho u_i u_j) dV$$

$$\begin{pmatrix} q_1 \\ q_2 \\ q_3 \end{pmatrix} \begin{pmatrix} u_1 \\ u_2 \\ u_3 \end{pmatrix} \begin{pmatrix} n_1 \\ n_2 \\ n_3 \end{pmatrix}$$

④

$$\vec{\Pi} = \int_V \vec{F}_{\text{body}} dV + \int_S \vec{F}_{\text{local}} dS$$

force densities

$$\Pi_i = \int_V F_{\text{body}_i} dV + \int_S F_{\text{local}_i} dS$$

$$F_{\text{body}_i} = \rho g_i + \dots$$

$$F_{\text{local}_i} = \tau_{ji} n_j \quad \longrightarrow \quad \int_S F_{\text{local}_i} dS = \int_S \tau_{ji} n_j dS$$

$$= \int_V \frac{\partial}{\partial x_j} \tau_{ji} dV \quad \left[\text{by the divergence theorem} \right]$$

$\bar{\bar{\tau}}$ = total stress tensor

$$\int_S \vec{F}_{\text{ext}} dS = \int_V \vec{\nabla} \cdot \bar{\bar{\tau}} dV$$

$$\Rightarrow \int_V \frac{\partial(\rho u_i)}{\partial t} dV = - \int_V \frac{\partial}{\partial x_j} (\rho u_i u_j) dV + \int_V \rho g_i dV + \int_V \frac{\partial}{\partial x_j} \tau_{ji} dV$$

$$\Rightarrow \frac{\partial(\rho u_i)}{\partial t} + \frac{\partial}{\partial x_j} (\rho u_i u_j) = \rho g_i + \frac{\partial}{\partial x_j} \tau_{ji} \quad \text{because we can take } V \text{ as small as the "blob" of fluid and therefore, the integrand has to vanish pointwise.}$$

$$\Rightarrow \rho \frac{\partial u_i}{\partial t} + \rho u_j \frac{\partial u_i}{\partial x_j} + \underbrace{u_i \frac{\partial \rho}{\partial t} + u_i \frac{\partial(\rho u_j)}{\partial x_j}}_{=0 \text{ because mass is conserved}} = \rho g_i + \frac{\partial}{\partial x_j} \tau_{ji}$$

$$\Rightarrow \rho \frac{\partial u_i}{\partial t} + \rho u_j \frac{\partial u_i}{\partial x_j} = \rho g_i + \frac{\partial \tau_{ji}}{\partial x_j}$$

$$\rho \frac{d\vec{u}}{dt} + \rho (\vec{u} \cdot \vec{\nabla}) \vec{u} = \rho \vec{g} + \vec{\nabla} \cdot \vec{\tau}$$

Equation of motion
conservation of momentum

Parenthesis on the velocity gradient: velocity gradient tensor $\bar{K}$

$$K_{ij} = \frac{\partial u_i}{\partial x_j}$$

$$= E_{ij}$$

symmetric

$$E_{ij} = E_{ji}$$

strain rate tensor

$$+ \Omega_{ij}$$

antisymmetric

$$\Omega_{ij} = -\Omega_{ji}$$

vorticity tensor

$$E_{ij} = \frac{1}{2} (K_{ij} + K_{ji})$$

$$\Omega_{ij} = \frac{1}{2} (K_{ij} - K_{ji})$$

Example in 2D: $\vec{u} = u \vec{e}_x + v \vec{e}_y$

incompressible $\partial_x u + \partial_y v = 0$

$$\bar{K} = \begin{pmatrix} \underline{\partial_x u} & \underline{\partial_y u} \\ \underline{\partial_x v} & -\underline{\partial_x u} \end{pmatrix}$$

3 unknowns

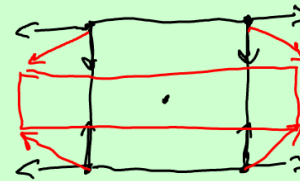
$$\bar{F} = \begin{pmatrix} a & b \\ b & -a \end{pmatrix}$$

$$\bar{\Sigma} = \begin{pmatrix} 0 & c \\ -c & 0 \end{pmatrix}$$

$$\bullet \bar{F}_a = \begin{pmatrix} 1 & 0 \\ 0 & -1 \end{pmatrix}$$

$$\partial_x u = 1$$

$$\partial_y v = -1$$

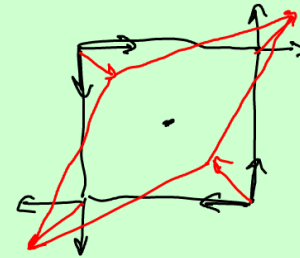


DEFORMATION

$$\bullet \bar{F}_b = \begin{pmatrix} 0 & 1 \\ 1 & 0 \end{pmatrix}$$

$$\partial_y u = 1$$

$$\partial_x v = 1$$

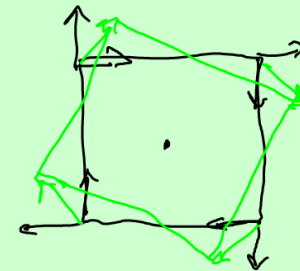


DEFORMATION

$$\bullet \bar{\Sigma}_c = \begin{pmatrix} 0 & 1 \\ -1 & 0 \end{pmatrix}$$

$$\partial_y u = 1$$

$$\partial_x v = -1$$



SOLID BODY
ROTATION

velocity gradient $\bar{K}$ $\begin{cases} \bar{E} & \text{symmetric, strain rate tensor} \rightarrow \text{deformations (force)} \\ \bar{\Omega} & \text{anti-symmetric, vorticity tensor} \rightarrow \text{solid body rotations (torques)} \end{cases}$

Vorticity: $\vec{\omega} = \vec{\nabla} \times \vec{u}$

$$\begin{aligned} \omega_i &= \varepsilon_{ijk} \frac{\partial}{\partial x_j} u_k \Rightarrow \varepsilon_{ilm} \omega_i = \varepsilon_{ilm} \varepsilon_{ijk} \frac{\partial}{\partial x_j} u_k \\ &= (\delta_{lj} \delta_{mk} - \delta_{lk} \delta_{mj}) \frac{\partial}{\partial x_j} u_k \\ &= \frac{\partial u_m}{\partial x_l} - \frac{\partial u_l}{\partial x_m} \\ &= 2 \Omega_{ml} \end{aligned}$$

$$\begin{aligned} \Rightarrow \Omega_{ml} &= \frac{1}{2} \varepsilon_{ilm} \omega_i \\ &= \frac{1}{2} \varepsilon_{klm} \omega_k \end{aligned}$$

$$\begin{aligned} \Rightarrow \Omega_{ij} &= \frac{1}{2} \varepsilon_{kji} \omega_k \\ &= -\frac{1}{2} \varepsilon_{ijk} \omega_k \end{aligned}$$

$$\bar{\Omega} = \begin{pmatrix} \vec{\omega} \end{pmatrix}$$

$$\rho \frac{d\vec{u}}{dt} + \rho (\vec{u} \cdot \vec{\nabla}) \vec{u} = \rho \vec{g} + \vec{\nabla} \cdot \vec{\bar{\sigma}} \quad \text{total stress tensor}$$

(5)

Constitutive equations:

* Ideal fluid: $\vec{F}_{\text{local}} = -P \vec{n}$
total pressure

$$\Rightarrow F_{\text{local } i} = \underbrace{\tau_{ji} n_j}_{\Rightarrow \tau_{ji} = -P \delta_{ij}} = -P n_i$$

$$\Rightarrow \tau_{ji} = -P \delta_{ij}$$

$$\vec{\bar{\sigma}} = \begin{pmatrix} -P & 0 & 0 \\ 0 & -P & 0 \\ 0 & 0 & -P \end{pmatrix}$$

$$\Rightarrow \rho \frac{\partial u_i}{\partial t} + \rho u_j \frac{\partial}{\partial x_j} u_i = \rho g_i + \frac{\partial}{\partial x_j} (-P \delta_{ji})$$

$$\Rightarrow \rho \frac{\partial u_i}{\partial t} + \rho u_j \frac{\partial}{\partial x_j} u_i = \rho g_i - \frac{\partial P}{\partial x_i}$$

$$\boxed{\rho \frac{d\vec{u}}{dt} + \rho (\vec{u} \cdot \vec{\nabla}) \vec{u} = \rho \vec{g} - \vec{\nabla} P}$$

+ incompressibility

Euler equation

- * Newtonian fluids:
 - fluid for which the relationship between strain rate and stress is linear
 - isotropic fluid

$$\rightarrow \tau_{ji} = -P \delta_{ij} + \sigma_{ji}$$

$\swarrow$ total stress tensor $\nearrow$ viscous stress tensor

$$\sigma_{ji} = 2\mu \left[\epsilon_{ij} - \frac{1}{3} \delta_{ij} \frac{\partial u_k}{\partial x_k} \right] + \dots$$

$\uparrow$ dynamic viscosity $\swarrow$ strain rate tensor $\nearrow$ incompressible fluid

— not to be confused with

$$\nu = \frac{\mu}{\rho} \quad \text{kinematic viscosity}$$

$$\sigma_{ji} = \mu \left(\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right)$$

$$\tau_{ji} = -P \delta_{ij} + \mu \left(\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right)$$

constant

$$\Rightarrow \rho \frac{\partial u_i}{\partial t} + \rho u_j \frac{\partial u_i}{\partial x_j} = \rho g_i - \frac{\partial P}{\partial x_i} + \mu \left(\frac{\partial^2 u_i}{\partial x_j^2} + \frac{\partial^2 u_j}{\partial x_i \partial x_j} \right) \rightarrow = 0 \text{ by incompressibility}$$

$$\Rightarrow \rho \frac{\partial u_i}{\partial t} + \rho u_j \frac{\partial u_i}{\partial x_j} = \rho g_i - \frac{\partial P}{\partial x_i} + \mu \frac{\partial^2 u_i}{\partial x_j^2}$$

$$\boxed{\rho \frac{\partial \vec{u}}{\partial t} + \rho (\vec{u} \cdot \vec{\nabla}) \vec{u} = \rho \vec{g} - \vec{\nabla} P + \mu \vec{\nabla}^2 \vec{u}} \quad \text{Navier-Stokes equation}$$

+ incompressibility ($\vec{\nabla} \cdot \vec{u} = 0$)

Different kinds of pressure: no flow, no forces $\rightarrow \vec{0} = \rho \vec{g} - \vec{\nabla} P_H$
(except the weight)

$$\Rightarrow P_H = \rho \vec{g} \cdot \vec{x} + P_0$$

hydrostatic pressure reference pressure

We can write $P = P_H + \uparrow$

total pressure dynamic pressure

|

hydrostatic pressure

The Navier-Stokes equation becomes:

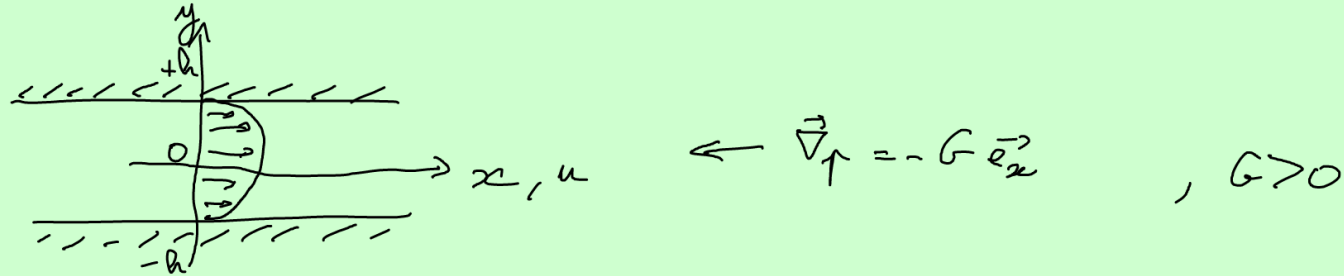
$$\boxed{\rho \frac{\partial \vec{u}}{\partial t} + \rho (\vec{u} \cdot \vec{\nabla}) \vec{u} = - \vec{\nabla} p + \mu \vec{\nabla}^2 \vec{u}}$$

Chapter 4: Some exact solutions of the Navier-Stokes equation

Plane Poiseuille flow

POISEUILLE

(0) Make a sketch



(1) Equations:

2D Navier-Stokes:

$$\begin{cases} \rho \frac{\partial u}{\partial t} + \rho u \frac{\partial u}{\partial x} + \rho v \frac{\partial u}{\partial y} = -\frac{\partial p}{\partial x} + \mu \frac{\partial^2 u}{\partial x^2} + \mu \frac{\partial^2 u}{\partial y^2} \\ \rho \frac{\partial v}{\partial t} + \rho u \frac{\partial v}{\partial x} + \rho v \frac{\partial v}{\partial y} = -\frac{\partial p}{\partial y} + \mu \frac{\partial^2 v}{\partial x^2} + \mu \frac{\partial^2 v}{\partial y^2} \end{cases}$$

Incompressibility:

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0$$

(2) Boundary conditions: no-slip:

$$\begin{cases} y = -h & u = 0, v = 0 \\ y = +h & u = 0, v = 0 \end{cases}$$

impermeability condition

(3) Hypotheses:

- * one-dimensional flow: $v \approx 0$
- * steady flow: $\frac{\partial}{\partial t} = 0$

(4) Solution:
$$\begin{cases} \rho \mu \frac{\partial u}{\partial x} = -\frac{\partial p}{\partial x} + \mu \frac{\partial^2 u}{\partial x^2} + \mu \frac{\partial^2 u}{\partial y^2} \\ 0 = -\frac{\partial p}{\partial y} \\ \frac{\partial u}{\partial x} = 0 \end{cases}$$

$$\begin{cases} 0 = -\frac{\partial p}{\partial x} + \mu \frac{\partial^2 u}{\partial y^2} \\ 0 = -\frac{\partial p}{\partial y} \\ \frac{\partial u}{\partial x} = 0 \end{cases} \rightarrow \begin{matrix} p(x) \\ u(y) \end{matrix}$$

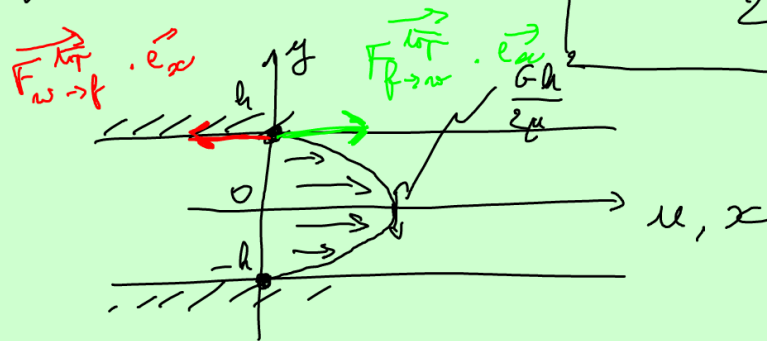
$$\overset{f(x)}{\frac{\partial p}{\partial x}} = \mu \overset{f(y)}{\frac{\partial^2 u}{\partial y^2}} = -G, \quad G > 0$$

$$\Rightarrow \frac{\partial^2 u}{\partial y^2} = -\frac{G}{\mu}$$

$$\Rightarrow u = -\frac{G}{2\mu} y^2 + k_1 y + k_2$$

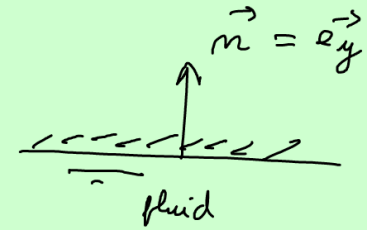
BC: $y = \pm h, u = 0$

$$u = \frac{G}{2\mu} (h^2 - y^2)$$



force density exerted by the wall onto the fluid at $y = +h$

$$\begin{aligned}\vec{F}_{w \rightarrow f}^{\text{tot}} &= \vec{n} \cdot \vec{\tau} \Big|_{y=+h} \\ &= \vec{e}_y \cdot \vec{\tau} \Big|_{y=+h}\end{aligned}$$

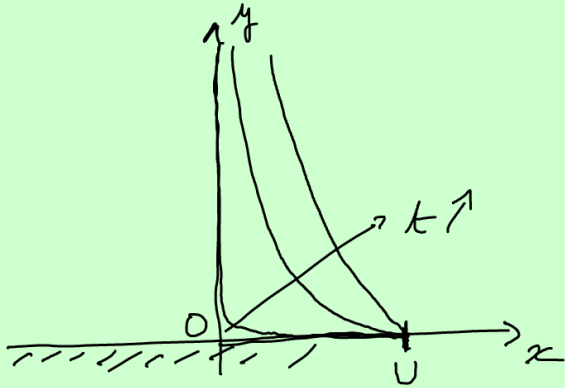


$$\begin{aligned}&= \tau_{yx} \Big|_{y=+h} \vec{e}_x + \tau_{yy} \Big|_{y=+h} \vec{e}_y \\ &= \mu \left(\frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \right) \Big|_{y=+h} \vec{e}_x + \left[-P + \mu \left(\frac{\partial v}{\partial y} + \frac{\partial v}{\partial y} \right) \right] \Big|_{y=+h} \vec{e}_y \\ &= \mu \frac{\partial u}{\partial y} \Big|_{y=+h} \vec{e}_x - P \vec{e}_y \\ &= \underline{-Gh} \vec{e}_x - P \vec{e}_y\end{aligned}$$

$$\vec{F}_{f \rightarrow w}^{\text{tot}} = - \vec{F}_{w \rightarrow f}^{\text{tot}}$$

Stokes first problem

(0)



$$(1) \begin{cases} \rho \frac{\partial u}{\partial t} + \rho u \frac{\partial u}{\partial x} + \cancel{\rho v \frac{\partial u}{\partial y}} = - \frac{\partial p}{\partial x} + \mu \frac{\partial^2 u}{\partial x^2} + \mu \frac{\partial^2 u}{\partial y^2} \\ \cancel{\rho \frac{\partial v}{\partial t}} + \cancel{\rho u \frac{\partial v}{\partial x}} + \cancel{\rho v \frac{\partial v}{\partial y}} = - \frac{\partial p}{\partial y} + \cancel{\mu \frac{\partial^2 v}{\partial x^2}} + \cancel{\mu \frac{\partial^2 v}{\partial y^2}} \\ \frac{\partial u}{\partial x} + \cancel{\frac{\partial v}{\partial y}} = 0 \end{cases}$$

$$(2) \text{ BC: } \begin{aligned} y=0 & , \quad u=U, \quad v=0 \\ y \rightarrow \infty & , \quad u \rightarrow 0, \quad v \rightarrow 0 \end{aligned}$$

$$\text{IC: } t=0, \quad u=0, \quad v=0$$

(3) Hypotheses: * one-dimensional: $v \equiv 0$ ✓
 + no force resulting from the dynamic pressure: $\vec{\nabla} p = 0$!

$$(4) \text{ Solution: } \begin{cases} \rho \frac{\partial u}{\partial t} + \cancel{\rho u \frac{\partial u}{\partial x}} = - \cancel{\frac{\partial p}{\partial x}} + \mu \cancel{\frac{\partial^2 u}{\partial x^2}} + \mu \frac{\partial^2 u}{\partial y^2} \\ 0 = - \frac{\partial p}{\partial y} \rightarrow p(x, t) \\ \frac{\partial u}{\partial x} = 0 \rightarrow u(y, t) \end{cases}$$

$$\Rightarrow \rho \frac{\partial u}{\partial t} = - \cancel{\frac{\partial p}{\partial x}} + \mu \frac{\partial^2 u}{\partial y^2}$$

$$\Rightarrow \boxed{\frac{\partial u}{\partial t} = \nu \frac{\partial^2 u}{\partial y^2}}$$

$$\nu = \frac{\mu}{\rho} \quad \text{kinematic viscosity}$$

heat equation
diffusion equation

2 methods: • separation of variables $u(y, t) = Y(y)T(t) \dots$
• similarity solution

similarity variable: $\eta = y t^a$

$$u(y, t) = f(\eta)$$

$$\partial_t = \partial_\eta \frac{\partial \eta}{\partial t} = a y t^{a-1} \partial_\eta$$

$$\partial_y = \partial_\eta \frac{\partial \eta}{\partial y} = t^a \partial_\eta$$

$$\partial_{yy} = \partial_y (\partial_y) = t^{2a} \partial_{\eta\eta}$$

$$\frac{\partial u}{\partial t} = \nu \frac{\partial^2 u}{\partial y^2}$$

$$\Rightarrow a y t^{a-1} f' = \nu t^{2a} f''$$

$$\Rightarrow f'' = \frac{a y t^{-a-1}}{\nu} f'$$

For this equation to be an equation of η only,
we need $t^a = t^{-a-1} \Rightarrow a = -\frac{1}{2}$

$$\Rightarrow \eta = y t^{-1/2} = \frac{y}{\sqrt{t}}$$

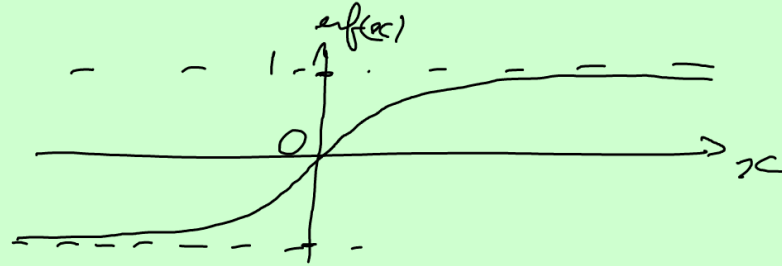
$$f'' = -\frac{\eta}{2\nu} f'$$

(7)

$$\Rightarrow f' = k_1 \exp\left(-\frac{\eta^2}{4v}\right)$$

$$\Rightarrow f = k_1 \underbrace{\int_0^{\eta} \exp\left(-\frac{\xi^2}{4v}\right) d\xi}_{\text{error function}} + k_2$$

Definition of the error function: $\text{erf}(x) = \frac{2}{\sqrt{\pi}} \int_0^x \exp(-y^2) dy$



$$y = \frac{\xi}{2\sqrt{v}} \quad \Rightarrow \quad \text{erf}(x) = \frac{2}{\sqrt{\pi}} \int_0^{2\sqrt{v}x} \exp\left(-\frac{\xi^2}{4v}\right) \frac{d\xi}{2\sqrt{v}}$$

$$= \frac{1}{\sqrt{v\pi}} \int_0^{2\sqrt{v}x} \exp\left(-\frac{\xi^2}{4v}\right) d\xi$$

$$\eta = 2\sqrt{v}x \quad \Rightarrow \quad \text{erf}\left(\frac{\eta}{2\sqrt{v}}\right) = \frac{1}{\sqrt{v\pi}} \int_0^{\eta} \exp\left(-\frac{\xi^2}{4v}\right) d\xi$$

$$\Rightarrow f = k_1 \sqrt{v\pi} \text{erf}\left(\frac{\eta}{2\sqrt{v}}\right) + k_2$$

$$\Rightarrow f = k_3 \operatorname{erf}\left(\frac{y}{2\sqrt{\nu t}}\right) + k_2$$

$$\Rightarrow u(y, t) = k_3 \operatorname{erf}\left(\frac{y}{2\sqrt{\nu t}}\right) + k_2$$

$$\text{BC: } y=0 \rightarrow u=U \Rightarrow k_2 = U$$

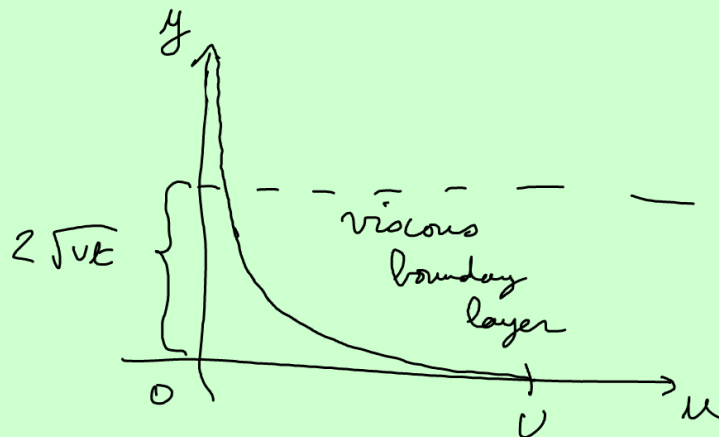
$$y \rightarrow \infty \rightarrow u \rightarrow 0 \Rightarrow k_3 + k_2 \rightarrow 0$$

$$\Rightarrow k_3 \rightarrow -U$$

$$\Rightarrow u(y, t) = U \left[1 - \operatorname{erf}\left(\frac{y}{2\sqrt{\nu t}}\right) \right]$$

$$\text{IC: } t \rightarrow 0 \quad u \rightarrow 0$$

this is compatible with the above expression
 $\Rightarrow$ the similarity condition is valid!

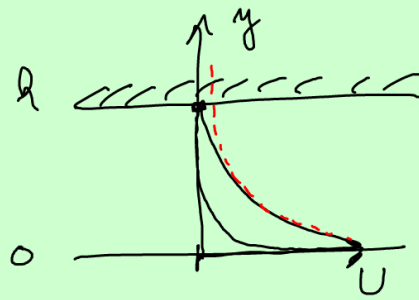


$$u \ll U \Rightarrow \frac{y}{2\sqrt{\nu t}} \gg 1$$

$$\Rightarrow y \gg 2\sqrt{\nu t}$$

$$\operatorname{erf}(2) \approx 0.99 \Rightarrow u \approx 0.01U \text{ for } y \geq \underbrace{4\sqrt{\nu t}}_{\delta_{99}}$$

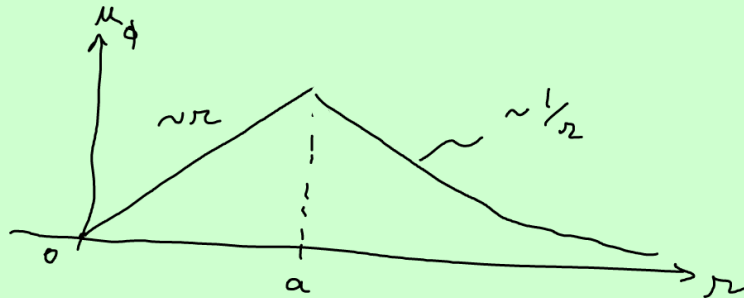
boundary layer thickness
 after the 99% criterion.



Chapter 5: Vorticity

$$\vec{\omega} = \vec{\nabla} \times \vec{u}$$

The Rankine vortex



$$\vec{u} = k r \vec{e}_\phi \quad \text{for } r \leq a$$

$$\vec{u} = \frac{k a^2}{r} \vec{e}_\phi \quad \text{for } r > a$$

vorticity: $\vec{\omega} = \vec{\nabla} \times \vec{u}$

$$= \begin{pmatrix} 0 \\ 0 \\ \frac{1}{r} \frac{d}{dr}(r u_\phi) \end{pmatrix}$$

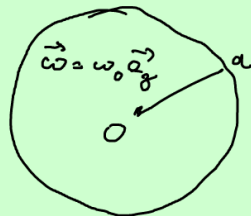
$$= \begin{cases} 2k \vec{e}_z & \text{for } r \leq a \\ \vec{0} & \text{for } r > a \end{cases}$$

$$\omega_0 = 2k$$

8



$$\vec{\omega} = \vec{0}$$



$$\text{Rankine vortex: } \vec{u} = \begin{cases} \frac{\omega_0 r}{2} \vec{e}_\phi & \text{for } r \leq a \\ \frac{\omega_0 a^2}{2r} \vec{e}_\phi & \text{for } r > a \end{cases}$$

Interesting calculation: calculate the pressure in the Rankine vortex

$$\vec{NS} \cdot \vec{e}_r : \quad \rho \frac{u_\phi^2}{r} = \frac{\partial p}{\partial r}$$

Line/point vortex:

$$\text{Around the Rankine vortex: } \int_S \vec{\omega} \cdot \vec{n} dS = \mp a^2 \omega_0$$

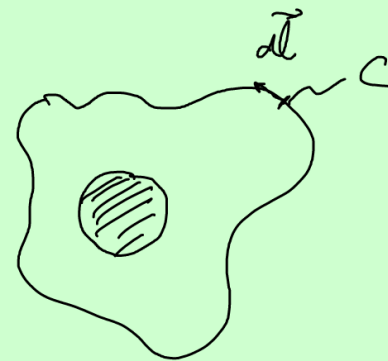
S — surface large enough
to contain the Rankine vortex

$$\text{By Stokes theorem: } \int_S \vec{\omega} \cdot \vec{n} dS = \oint_C \vec{u} \cdot d\vec{l}$$

C
closed contour corresponding to S

$$\Rightarrow \mp a^2 \omega_0 = \oint_C \vec{u} \cdot d\vec{l} = \Gamma$$

$\uparrow$
circulation



Taking α to 0 while maintaining Γ constant (increasing ω_0 in proportion) leads to the creation of a point vortex (line vortices in 3D)

↳ potential flow theory

The vorticity equation

$$\rho \left[\frac{d\vec{u}}{dt} + (\vec{u} \cdot \vec{\nabla}) \vec{u} \right] = -\vec{\nabla} p + \mu \vec{\nabla}^2 \vec{u} + \vec{F}$$

$$\begin{aligned} \underline{\vec{u} \times (\vec{\nabla} \times \vec{u})}_i &= \varepsilon_{ijk} u_j \underline{(\vec{\nabla} \times \vec{u})}_k \\ &= \varepsilon_{ijk} u_j \varepsilon_{klm} \frac{\partial}{\partial x_l} u_m \\ &= \varepsilon_{ijk} \varepsilon_{klm} u_j \frac{\partial}{\partial x_l} u_m \\ &= \varepsilon_{kij} \varepsilon_{klm} u_j \frac{\partial}{\partial x_l} u_m \\ &= (\delta_{il} \delta_{jm} - \delta_{im} \delta_{jl}) u_j \frac{\partial}{\partial x_l} u_m \\ &= u_j \frac{\partial}{\partial x_i} u_j - u_j \frac{\partial}{\partial x_j} u_i \end{aligned}$$

$$= \frac{1}{2} \frac{\partial}{\partial x_i} u_j^2 - u_j \frac{\partial}{\partial x_j} u_i$$

$$\Rightarrow \vec{u} \times (\vec{\nabla} \times \vec{u}) = \frac{1}{2} \vec{\nabla} |\vec{u}|^2 - (\vec{u} \cdot \vec{\nabla}) \vec{u}$$

$$\hookrightarrow \rho \left[\frac{\partial \vec{u}}{\partial t} + \frac{1}{2} \vec{\nabla} |\vec{u}|^2 - \vec{u} \times \vec{\omega} \right] = -\vec{\nabla} p + \mu \vec{\nabla}^2 \vec{u} + \vec{F}$$

$$\Rightarrow \rho \left[\frac{\partial \vec{u}}{\partial t} - \vec{u} \times \vec{\omega} \right] = -\vec{\nabla} \left(p + \frac{1}{2} \rho |\vec{u}|^2 \right) + \mu \vec{\nabla}^2 \vec{u} + \vec{F}$$

We take the curl:

$$\Rightarrow \rho \left[\frac{\partial \vec{\omega}}{\partial t} - \vec{\nabla} \times (\vec{u} \times \vec{\omega}) \right] = \vec{0} + \mu \vec{\nabla}^2 \vec{\omega} + \vec{0} \leftarrow \text{for } \vec{F} = \vec{\nabla} V \text{ force deriving from a potential}$$

$$\vec{\nabla} \times (\vec{u} \times \vec{\omega}) \Big|_i = \varepsilon_{ijk} \frac{\partial}{\partial x_j} \left(\varepsilon_{klm} u_l \omega_m \right)$$

$$= \varepsilon_{kij} \varepsilon_{klm} \frac{\partial}{\partial x_j} (u_l \omega_m)$$

$$= (\delta_{il} \delta_{jm} - \delta_{im} \delta_{jl}) \frac{\partial}{\partial x_j} (u_l \omega_m)$$

$$= \frac{\partial}{\partial x_j} (\mu_i \omega_{ji}) - \frac{\partial}{\partial x_j} (\mu_j \omega_{ji})$$

$$= \mu_i \cancel{\frac{\partial \omega_{ji}}{\partial x_j}} + \omega_{ji} \frac{\partial \mu_i}{\partial x_j} - \mu_j \frac{\partial \omega_{ji}}{\partial x_j} - \omega_{ji} \cancel{\frac{\partial \mu_j}{\partial x_j}} \quad \text{incompressible}$$

incompressible

$$= \omega_{ji} \frac{\partial \mu_i}{\partial x_j} - \mu_j \frac{\partial \omega_{ji}}{\partial x_j}$$

$$\Rightarrow \vec{\nabla} \times (\vec{u} \times \vec{\omega}) = (\vec{\omega} \cdot \vec{\nabla}) \vec{u} - (\vec{u} \cdot \vec{\nabla}) \vec{\omega}$$

$$\hookrightarrow \rho \left[\frac{\partial \vec{\omega}}{\partial t} + (\vec{u} \cdot \vec{\nabla}) \vec{\omega} - (\vec{\omega} \cdot \vec{\nabla}) \vec{u} \right] = \mu \vec{\nabla}^2 \vec{\omega}$$

$$\Rightarrow \boxed{\frac{\partial \vec{\omega}}{\partial t} + (\vec{u} \cdot \vec{\nabla}) \vec{\omega} = (\vec{\omega} \cdot \vec{\nabla}) \vec{u} + \nu \vec{\nabla}^2 \vec{\omega}}$$

rate of change

advection of vorticity

vortex stretching

diffusion

Vorticity equation

To understand $(\vec{\omega} \cdot \vec{\nabla}) \vec{u}$, we isolate it: $\frac{\partial \vec{\omega}}{\partial t} = (\vec{\omega} \cdot \vec{\nabla}) \vec{u}$

Align the z -axis with $\vec{\omega}$: $\vec{\omega} = \omega \vec{e}_z$

$$\Rightarrow \frac{\partial \omega}{\partial t} = (\vec{\omega} \cdot \vec{\nabla}) u_z$$

$$= \omega \frac{\partial u_z}{\partial z}$$

$$\Rightarrow \frac{1}{\omega} \frac{\partial \omega}{\partial t} = \frac{\partial u_z}{\partial z}$$

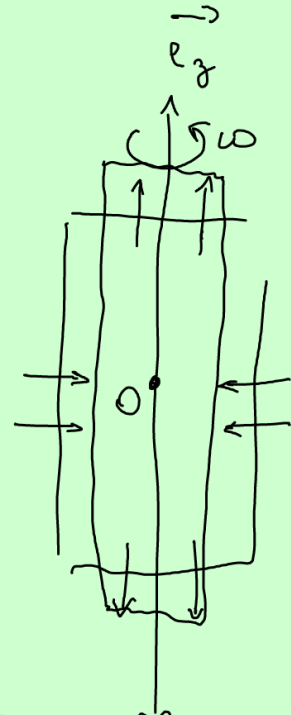
$$\text{if } \frac{\partial u_z}{\partial z} > 0, \quad \frac{\partial u_r}{\partial r} < 0$$

we are sucking in fluid radially

to compensate for the z -expansion

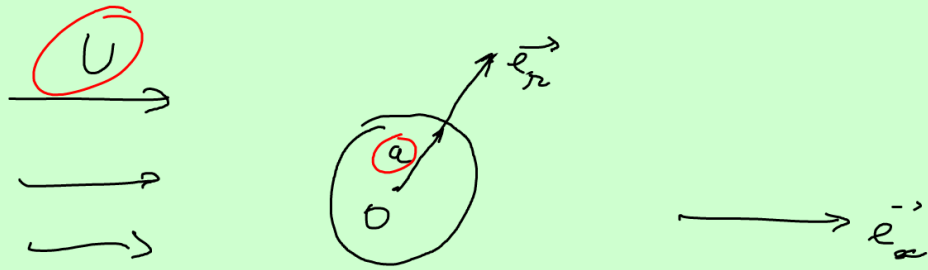
$\Rightarrow$ stretching of the flow in z .

$$\frac{1}{\omega} \frac{\partial \omega}{\partial t} > 0 \Rightarrow |\omega| \nearrow$$



Chapter 6: Dynamic similarity

9



μ : dynamic viscosity
 ρ : density

$$\text{Navier-Stokes: } \rho \left[\frac{d\vec{u}}{dt} + (\vec{u} \cdot \vec{\nabla}) \vec{u} \right] = - \vec{\nabla} p + \mu \vec{\nabla}^2 \vec{u}$$

$$\text{incompressibility: } \vec{\nabla} \cdot \vec{u} = 0$$

$$\text{BC, no-slip: } \begin{cases} \vec{u}(r=a) = \vec{0} \\ \lim_{r \rightarrow \infty} \vec{u} = U \vec{e}_\infty \end{cases}$$

IC or look for a steady state ($\leftarrow$ a strong hypothesis?)

$\Rightarrow$ 4 parameters: μ, ρ, a, U

Non-dimensionalization:

$$\vec{x} = a \vec{x}^*$$

$$\hookrightarrow [x] = L$$

$$[a] = L$$

$$[x^*] = \frac{[x]}{[a]} = \frac{L}{L} = 1$$

$$x = 3.2 \text{ m}$$

$$a = 2 \text{ m}$$

$$x^* = 1.6$$

$$\vec{u} = U \vec{u}^*$$

$$\hookrightarrow [u] = L \cdot T^{-1}$$

$$[U] = L \cdot T^{-1}$$

$$[u^*] = 1$$

$$u = 700 \text{ m} \cdot \text{s}^{-1}$$

$$U = 1000 \text{ m} \cdot \text{s}^{-1}$$

$$u^* = 0.7$$

$$t = \frac{a}{U} t^*$$

$$\frac{[a]}{[U]} = T$$

$$p = P p^*$$

$$[p^*] = 1$$

↑ not the total pressure but a pressure scale

Incompressibility: $\vec{\nabla} \cdot \vec{u} = 0 \Rightarrow \frac{1}{a} \vec{\nabla}^* \cdot (U \vec{u}^*) = 0$

$$\Rightarrow \frac{U}{a} \vec{\nabla}^* \cdot \vec{u}^* = 0$$

$$\Rightarrow \vec{\nabla}^* \cdot \vec{u}^* = 0$$

Navier-Stokes:

$$\rho \left[\frac{U}{a} \frac{d}{dt^*} (U \vec{u}^*) + \frac{U^2}{a} (\vec{u}^* \cdot \vec{\nabla}^*) \vec{u}^* \right] = - \frac{P}{a} \vec{\nabla}^* p^* + \mu \frac{U}{a^2} \vec{\nabla}^{*2} \vec{u}^*$$

$$\Rightarrow \rho \frac{U^2}{a} \left[\frac{d \vec{u}^*}{dt^*} + (\vec{u}^* \cdot \vec{\nabla}^*) \vec{u}^* \right] = - \frac{P}{a} \vec{\nabla}^* p^* + \frac{\mu U}{a^2} \vec{\nabla}^{*2} \vec{u}^*$$

$$\Rightarrow \frac{d \vec{u}^*}{dt^*} + (\vec{u}^* \cdot \vec{\nabla}^*) \vec{u}^* = - \frac{P}{\rho U^2} \vec{\nabla}^* p^* + \frac{\mu}{\rho U a} \vec{\nabla}^{*2} \vec{u}^*$$

$$[P] = M \cdot L^{-1} \cdot T^{-2}$$

$$\left. \begin{array}{l} [\rho] = M \cdot L^{-3} \\ [U^2] = L^2 \cdot T^{-2} \end{array} \right\} [\rho U^2] = M \cdot L^{-1} \cdot T^{-2}$$

$Re = \frac{\rho U a}{\mu}$: non-dimensional number quantifying the ratio between inertial and viscous effects.

$$\Rightarrow \frac{d \vec{u}^*}{dt^*} + (\vec{u}^* \cdot \vec{\nabla}^*) \vec{u}^* = - \frac{1}{\rho U^2} \vec{\nabla}^* p^* + \frac{1}{Re} \vec{\nabla}^{*2} \vec{u}^*$$

Two choices for P : • $P = \rho U^2$ inertial scale

$$\hookrightarrow \frac{\partial \vec{u}^*}{\partial t^*} + (\vec{u}^* \cdot \vec{\nabla}^*) \vec{u}^* = - \vec{\nabla}^* p^* + \frac{1}{\text{Re}} \vec{\nabla}^{*2} \vec{u}^*$$

• $P = \frac{\mu U}{a}$ viscous scale

$$\text{Re} \left[\frac{\partial \vec{u}^*}{\partial t^*} + (\vec{u}^* \cdot \vec{\nabla}^*) \vec{u}^* \right] = - \vec{\nabla}^* p^* + \vec{\nabla}^{*2} \vec{u}^*$$

$$\text{BC: } r^* \rightarrow \infty, \quad \vec{u}^* = \vec{e}_x'$$

$$r^* = 1, \quad \vec{u}^* = \vec{0}$$

$$\text{Re} = \rho \frac{U a'}{\mu} = 100$$

$$\text{say: } U a = \frac{\text{Re} \mu}{\rho}$$

Chapter 7: Potential flows

(10)

Two dimensional flow, irrotational, incompressible

$$\vec{u}(x, y) = u(x, y) \vec{e}_x + v(x, y) \vec{e}_y$$

$$\vec{\nabla} \times \vec{u} = \vec{0}$$

$$\vec{\nabla} \cdot \vec{u} = 0$$

First route: (1) incompressible: $\vec{\nabla} \cdot \vec{u} = 0$

↳ streamfunction ψ : $u = \frac{\partial \psi}{\partial y}$, $v = -\frac{\partial \psi}{\partial x}$

$$\Rightarrow \vec{\nabla} \cdot \vec{u} = \frac{\partial^2 \psi}{\partial x \partial y} - \frac{\partial^2 \psi}{\partial x \partial y} = 0$$

$$(2) \text{ irrotational: } \vec{\nabla} \times \vec{u} = \left(\frac{\partial v}{\partial x} - \frac{\partial u}{\partial y} \right) \vec{e}_z$$

$$= \left(-\frac{\partial^2 \psi}{\partial x^2} - \frac{\partial^2 \psi}{\partial y^2} \right) \vec{e}_z$$

$$= \vec{0}$$

$$\Rightarrow \frac{\partial^2 \psi}{\partial x^2} + \frac{\partial^2 \psi}{\partial y^2} = 0$$

$$\Rightarrow \boxed{\vec{\nabla}^2 \psi = 0}$$

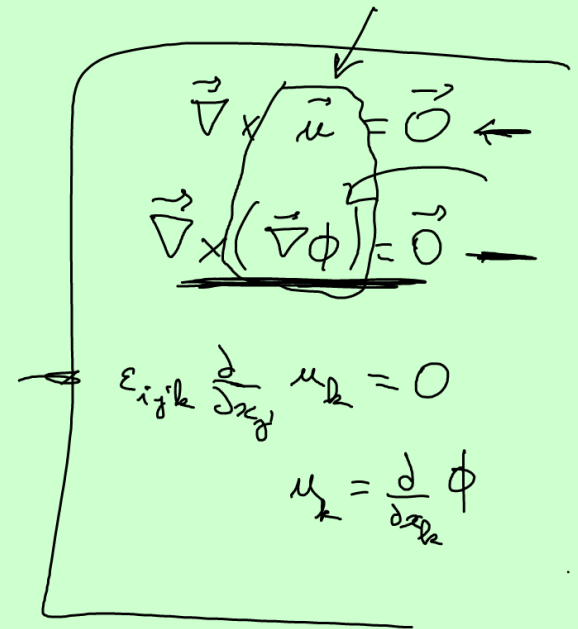


Diagram illustrating the relationship between velocity vectors $\vec{u}$ and $\vec{v}$ and the streamfunction ψ . The diagram shows a fluid element with velocity components u and v and the corresponding streamfunction derivatives $\frac{\partial \psi}{\partial y}$ and $-\frac{\partial \psi}{\partial x}$. The curl of the velocity field is shown to be zero, indicating irrotational flow.

$$\vec{\nabla} \times \vec{u} = \vec{0}$$

$$\vec{\nabla} \times (\vec{\nabla} \psi) = \vec{0}$$

$$\epsilon_{ijk} \frac{\partial}{\partial x_j} u_k = 0$$

$$u_k = \frac{\partial \psi}{\partial x_k}$$

Second route: (1) Irrotational: $\vec{\nabla} \times \vec{u} = \vec{0}$

↳ velocity potential ϕ : $u = \frac{\partial \phi}{\partial x}$, $v = \frac{\partial \phi}{\partial y}$

$$\Rightarrow \vec{\nabla} \times \vec{u} = \left(\frac{\partial v}{\partial x} - \frac{\partial u}{\partial y} \right) \vec{e}_z$$

$$= \left(\frac{\partial^2 \phi}{\partial x \partial y} - \frac{\partial^2 \phi}{\partial x \partial y} \right) \vec{e}_z = \vec{0}$$

(2) Incompressible: $\vec{\nabla} \cdot \vec{u} = 0 \Rightarrow \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0$

$$\Rightarrow \frac{\partial^2 \phi}{\partial x^2} + \frac{\partial^2 \phi}{\partial y^2} = 0$$

$$\Rightarrow \boxed{\vec{\nabla}^2 \phi = 0}$$

The velocity is:

$$u = \frac{\partial \phi}{\partial x} = \frac{\partial \psi}{\partial y}$$

$$v = \frac{\partial \phi}{\partial y} = -\frac{\partial \psi}{\partial x}$$

Cauchy - Riemann equations

$$\left. \begin{aligned} \frac{\partial^2 \phi}{\partial x^2} &= \frac{\partial^2 \psi}{\partial x \partial y} \\ \frac{\partial^2 \phi}{\partial y^2} &= -\frac{\partial^2 \psi}{\partial x \partial y} \end{aligned} \right\} \Rightarrow \vec{\nabla}^2 \phi = 0$$

$$\left. \begin{aligned} \frac{\partial^2 \phi}{\partial x \partial y} &= \frac{\partial^2 \psi}{\partial y^2} \\ \frac{\partial^2 \phi}{\partial x \partial y} &= -\frac{\partial^2 \psi}{\partial x^2} \end{aligned} \right\} \Rightarrow \vec{\nabla}^2 \psi = 0$$

Complex analysis: $\frac{df}{dz} = \lim_{\delta z \rightarrow 0} \frac{f(x+\delta x) - f(x)}{\delta x}$

$$\frac{df}{dz} = \lim_{|\delta z| \rightarrow 0} \frac{f(z+\delta z) - f(z)}{\delta z}$$

$$z = x + iy$$

has to be the same no matter the direction of δz for f to be complex differentiable.

$$f: \mathbb{C} \mapsto \mathbb{C}$$

$$f(z) = g(x, y) + i h(x, y)$$

$$g: \mathbb{R}^2 \mapsto \mathbb{R}$$

$$h: \mathbb{R}^2 \mapsto \mathbb{R}$$

$$\text{In the } \delta x \text{ direction : } \frac{df}{dz} =$$

$$\frac{\partial g}{\partial x} + i \frac{\partial h}{\partial x}$$

$$\text{In the } i\delta y \text{ direction : } \frac{df}{dz} = \frac{\partial g}{i\delta y} + i \frac{\partial h}{i\delta y} = \frac{\partial h}{\partial y} - i \frac{\partial g}{\partial y}$$

$$\text{For } f \text{ to be complex differentiable : } \frac{\partial g}{\partial x} + i \frac{\partial h}{\partial x} = \frac{\partial h}{\partial y} - i \frac{\partial g}{\partial y}$$

$$\Rightarrow \begin{cases} \frac{\partial g}{\partial x} = \frac{\partial h}{\partial y} \\ \frac{\partial h}{\partial x} = -\frac{\partial g}{\partial y} \end{cases}$$

Cauchy - Riemann
equations

$$(\Rightarrow \vec{\nabla}^2 g = \vec{\nabla}^2 h = 0)$$

f is complex differentiable $\Rightarrow$ its real and imaginary parts are harmonic functions.

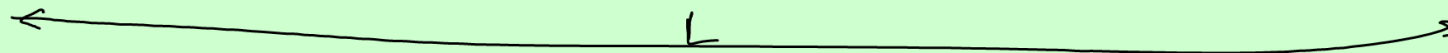
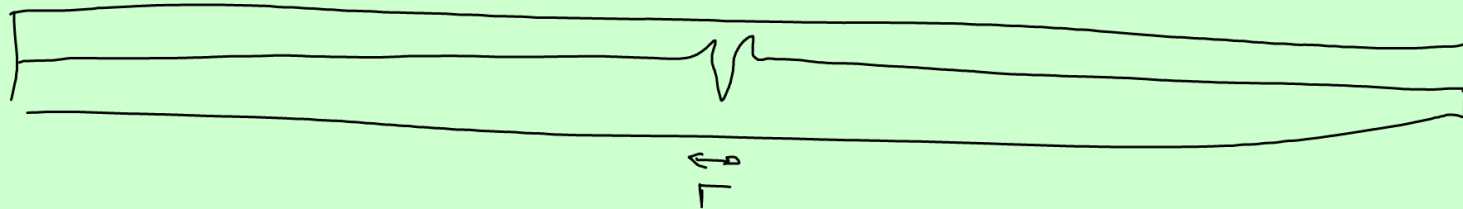
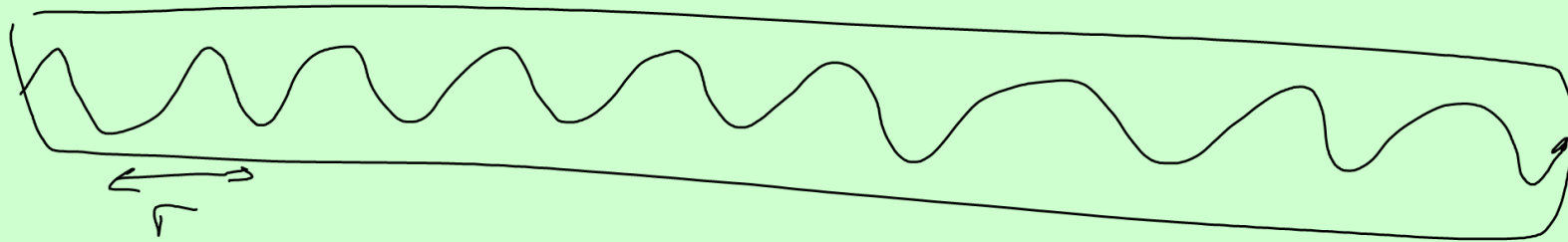
We can then introduce a complex potential: $w = \phi + i\psi$

$\swarrow$ complex potential
 $\searrow$ streamfunction
 $\swarrow$ velocity potential

w is complex differentiable $\Rightarrow \phi$ and ψ are harmonic functions
 $(\nabla^2 \phi = \nabla^2 \psi = 0)$
 incompressible and irrotational

$$\begin{aligned} \frac{dw}{dz} &= \frac{\partial \phi}{\partial x} + i \frac{\partial \psi}{\partial x} = u - i v \\ &= \frac{\partial \phi}{i \partial y} + i \frac{\partial \psi}{i \partial y} = \frac{\partial \psi}{\partial y} - i \frac{\partial \phi}{\partial y} = u - i v \end{aligned}$$

$\swarrow$ complex conjugate
 $u + i v = \overline{\frac{dw}{dz}}$



$x = \varepsilon X, \quad \varepsilon \ll 1$

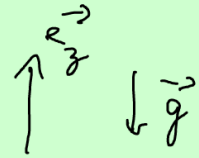
$\frac{d}{dx} = \frac{\partial}{\partial x} + \varepsilon \frac{\partial}{\partial X}$

The Bernoulli equation

(11)

Inviscid flow $\rightarrow$ Euler equation

$$\rho \left[\frac{\partial \vec{u}}{\partial t} + (\vec{u} \cdot \vec{\nabla}) \vec{u} \right] = - \vec{\nabla} P - \rho g \vec{e}_z$$



$$\Rightarrow \rho \left[\frac{\partial \vec{u}}{\partial t} - \vec{u} \times \vec{\omega} + \frac{1}{2} \vec{\nabla} |\vec{u}|^2 \right] = - \vec{\nabla} P - \rho g \vec{e}_z$$

$$\Rightarrow \rho \left[\frac{\partial \vec{u}}{\partial t} - \vec{u} \times \vec{\omega} \right] = - \vec{\nabla} \left(P + \frac{1}{2} \rho |\vec{u}|^2 \right) - \rho g \vec{e}_z$$

$$\Rightarrow \boxed{\rho \left[\frac{\partial \vec{u}}{\partial t} - \vec{u} \times \vec{\omega} \right] = - \vec{\nabla} \left(P + \frac{1}{2} \rho |\vec{u}|^2 + \rho g z \right)}$$

Case 1: steady, irrotational

$$\vec{\nabla} \left(P + \frac{1}{2} \rho |\vec{u}|^2 + \rho g z \right) = \vec{0}$$

$$\Rightarrow P + \frac{1}{2} \rho |\vec{u}|^2 + \rho g z = h_1(A)$$

$$\Rightarrow \boxed{P + \frac{1}{2} \rho |\vec{u}|^2 + \rho g z = h_2} \quad \text{because the flow is steady}$$

constant everywhere and at every time

Case 2: steady

$$\rho \vec{u} \times \vec{\omega} = \vec{\nabla} \left(P + \frac{1}{2} \rho |\vec{u}|^2 + \rho g z \right)$$

$$\Rightarrow \rho \vec{u} \cdot (\vec{u} \times \vec{\omega}) = \vec{u} \cdot \vec{\nabla} \left(P + \frac{1}{2} \rho |\vec{u}|^2 + \rho g z \right)$$

$$\Rightarrow 0 = \vec{u} \cdot \vec{\nabla} \left(P + \frac{1}{2} \rho |\vec{u}|^2 + \rho g z \right)$$

$$\Rightarrow \boxed{P + \frac{1}{2} \rho |\vec{u}|^2 + \rho g z = h_3(\psi)}$$

← constant along a streamline
(varies with the streamfunction value)

Case 3: irrotational

$$\rho \frac{\partial \vec{u}}{\partial t} = - \vec{\nabla} \left(P + \frac{1}{2} \rho |\vec{u}|^2 + \rho g z \right)$$

$$\Rightarrow \rho \frac{\partial \vec{\nabla} \phi}{\partial t} = - \vec{\nabla} \left(P + \frac{1}{2} \rho |\vec{u}|^2 + \rho g z \right)$$

$$\Rightarrow \vec{\nabla} \left(\rho \frac{\partial \phi}{\partial t} + P + \frac{1}{2} \rho |\vec{u}|^2 + \rho g z \right) = \vec{0}$$

$$\Rightarrow \boxed{\rho \frac{\partial \phi}{\partial t} + P + \frac{1}{2} \rho |\vec{u}|^2 + \rho g z = h_4(t)}$$

← constant that can vary with time

Lecture 19: High Re flows

9W

$$\frac{\partial \vec{u}^*}{\partial t^*} + (\vec{u}^* \cdot \vec{\nabla}^*) \vec{u}^* = -\vec{\nabla}^* p^* + \underbrace{\frac{1}{Re} \vec{\nabla}^{*2} \vec{u}^*}_{\substack{\text{do not neglect!} \\ \text{singular perturbation.}}}$$

Boundary layer will develop near structures.
Away from the boundary layer, inertia dominates

Inside the boundary layer: $\rho (\vec{u} \cdot \vec{\nabla}) \vec{u} \sim \mu \vec{\nabla}^2 \vec{u}$

$$\Rightarrow \underbrace{\frac{\rho U^2}{L}}_{\substack{\text{inertial scale} \\ \text{advective}}} \sim \underbrace{\frac{\mu U}{\delta^2}}_{\text{boundary layer thickness}}$$

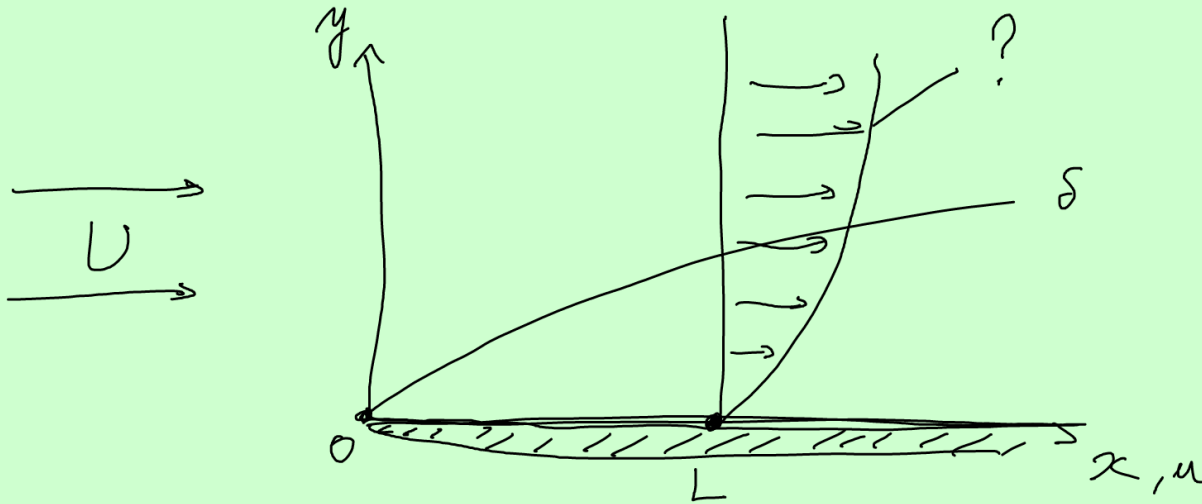
$$\Rightarrow \delta \sim \sqrt{\frac{\mu L}{\rho U}}$$

$$\Rightarrow \frac{\delta}{L} \sim Re_L^{-1/2}$$

$$Re_L = \frac{\rho U L}{\mu}$$

The Blasius boundary layer

PRANDTL & BLASIUS



- Hypotheses :
- Steady
 - 2D
 - large Re
 - thin boundary layer $\varepsilon = \frac{\delta}{L} \ll 1$

Equations :

$$\begin{cases} \rho u \frac{\partial u}{\partial x} + \rho v \frac{\partial u}{\partial y} = -\frac{\partial p}{\partial x} + \mu \frac{\partial^2 u}{\partial x^2} + \mu \frac{\partial^2 u}{\partial y^2} \\ \rho u \frac{\partial v}{\partial x} + \rho v \frac{\partial v}{\partial y} = -\frac{\partial p}{\partial y} + \mu \frac{\partial^2 v}{\partial x^2} + \mu \frac{\partial^2 v}{\partial y^2} \\ \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0 \end{cases}$$

3 unknowns : u, v, p

2D : x, y

BC: $y=0$, $u=0$, $v=0$
 $y \rightarrow \infty$, $u \rightarrow U$, $v \rightarrow 0$

Non-dimensionalization:
$$\begin{cases} x = L x^* \\ y = \delta y^* = \varepsilon L y^* \\ u = U u^* \\ v = \varepsilon U v^* \\ p = \rho U^2 p^* \end{cases}$$

$$\begin{cases} \frac{\rho U^2}{L} \left[u^* \partial_{x^*} u^* + v^* \partial_{y^*} u^* \right] = - \frac{\rho U^2}{L} \partial_{x^*} p^* + \mu \left[\frac{U}{L^2} \partial_{x^*}^2 u^* + \frac{U}{\varepsilon^2 L^2} \partial_{y^*}^2 u^* \right] \\ \frac{\rho \varepsilon U^2}{L} \left[u^* \partial_{x^*} v^* + v^* \partial_{y^*} v^* \right] = - \frac{\rho U^2}{\varepsilon L} \partial_{y^*} p^* + \mu \left[\frac{U \varepsilon}{L^2} \partial_{x^*}^2 v^* + \frac{U}{\varepsilon L^2} \partial_{y^*}^2 v^* \right] \\ \partial_{x^*} u^* + \partial_{y^*} v^* = 0 \end{cases}$$

$\Rightarrow$
$$\begin{cases} u^* \partial_{x^*} u^* + v^* \partial_{y^*} u^* = - \partial_{x^*} p^* + \frac{\mu}{\rho U L} \partial_{x^*}^2 u^* + \frac{\mu \varepsilon^{-2}}{\rho U L} \partial_{y^*}^2 u^* \\ \cancel{u^* \partial_{x^*} v^* + v^* \partial_{y^*} v^*} = - \varepsilon^{-2} \partial_{y^*} p^* + \frac{\mu}{\rho U L} \cancel{\partial_{x^*}^2 v^*} + \frac{\mu \varepsilon^{-2}}{\rho U L} \partial_{y^*}^2 v^* \end{cases}$$

smaller than $\downarrow$

$\leftarrow O(\varepsilon^{-2})$

$$\begin{cases} \partial_x u^* + \partial_y v^* = 0 \end{cases}$$

↑ smaller than ↑

$$\Rightarrow \begin{cases} u^* \partial_x u^* + v^* \partial_y u^* = -\partial_x p^* + \frac{\varepsilon^{-2}}{Re_L} \partial_y^2 u^* \\ 0 = -\varepsilon^{-2} \partial_y p^* + \frac{\varepsilon^{-2}}{Re_L} \partial_y^2 v^* \\ \partial_x u^* + \partial_y v^* = 0 \end{cases} \quad Re_L = \frac{\rho UL}{\mu}$$

small because $Re_L \gg 1$
compared to the $\partial_y p$ term

(M10)

To preserve the balance between advection (inertia) and diffusion in the boundary layer, we have to set $Re_L = \varepsilon^{-2}$

$$\left(\Rightarrow \frac{\rho UL}{\mu} = \varepsilon^{-2} = \frac{L^2}{\delta^2} \right)$$

$$\Rightarrow \delta = \sqrt{\frac{L\mu}{\rho U}}$$

$$\Rightarrow \begin{cases} \rho u \partial_x u + \rho v \partial_y u = -\partial_x p + \mu \partial_y^2 u \\ \partial_y p = 0 \\ \partial_x u + \partial_y v = 0 \end{cases}$$

Since $\partial_y p = 0$, we can "read" the pressure inside the boundary layer by taking its value infinitely far away from it but at the same value of x .

On a $y \rightarrow \infty$ streamline : $p + \frac{1}{2} \rho |\mathbf{U}|^2 = \text{cst} \Rightarrow p + \frac{1}{2} \rho U^2 = \text{cst}$
 $\Rightarrow p = \text{cst}$
 $\Rightarrow \partial_x p = 0$

$$\Rightarrow \begin{cases} \rho u \partial_x u + \rho v \partial_y u = \mu \partial_y^2 u \\ \partial_x u + \partial_y v = 0 \end{cases}$$

2 unknowns : u, v
 2D : x, y

Because the flow is 2D and incompressible, we can introduce the streamfunction ψ :

$$u = \partial_y \psi \quad v = -\partial_x \psi$$

$$\Rightarrow \partial_y \psi \partial_{xy} \psi - \partial_x \psi \partial_{yy} \psi = \nu \partial_{yyy} \psi$$

, $\nu = \frac{\mu}{\rho}$ the kinematic viscosity

$$\boxed{\frac{\partial \psi}{\partial y} \frac{\partial^2 \psi}{\partial x \partial y} - \frac{\partial \psi}{\partial x} \frac{\partial^2 \psi}{\partial y^2} = \nu \frac{\partial^3 \psi}{\partial y^3}}$$

1 unknown : ψ
 2D : x, y

Blasius: the boundary layer is self-similar: • $\eta = y \left(\frac{U}{\nu x} \right)^{1/2}$

$$= \frac{y}{f(x)}$$

$$f(x) = \left(\frac{\nu x}{U} \right)^{1/2}$$

$$\bullet u = U f'$$

$$f' = \frac{df}{d\eta}$$

$$u = \frac{\partial \psi}{\partial y} \Rightarrow U f' = \frac{\partial \psi}{\partial \eta} \frac{\partial \eta}{\partial y}$$

$$= \frac{\partial \psi}{\partial \eta} \frac{1}{f}$$

$$\Rightarrow \frac{\partial \psi}{\partial \eta} = U f f'$$

$$\Rightarrow \psi = U f f$$

$$\partial_x \psi = U f' f + U f \frac{\partial f}{\partial \eta} \frac{\partial \eta}{\partial x}$$

$$= U f' f - \frac{U y f' f'}{f} = -v$$

$$\partial_y \psi = U f f' \frac{\partial \eta}{\partial y} \quad \left| \quad \partial_{yy} \psi = \partial_y (U f f') \right.$$

$$= U f f' \quad \left| \quad = U f'' \frac{\partial \eta}{\partial y} \right.$$

$$= u \quad \left| \quad = \frac{U f''}{f} \right.$$

$$\frac{\partial \eta}{\partial x} = - \frac{y f'}{f^2}$$

$$\partial_{yyy} \psi = \frac{U f'''}{f^2}$$

$$\begin{aligned}\partial_{xy} \psi &= \partial_x (U f') \\ &= U f'' \frac{\partial y}{\partial x} \\ &= - \frac{U f'' y \delta'}{\delta^2}\end{aligned}$$

$$\Rightarrow U f' \left[- \frac{U y \delta'}{\delta^2} f'' \right] - \underbrace{\left(U \delta' f - \frac{U y \delta'}{\delta} f' \right)}_{\frac{U f''}{\delta}} = v \frac{U f'''}{\delta^2}$$

$$\Rightarrow - \frac{U^2 y \delta'}{\delta^2} f' f'' - \frac{U^2 \delta'}{\delta} f f'' + \frac{U^2 y \delta'}{\delta^2} f' f'' = \frac{v}{\delta^2} U f'''$$

$$\Rightarrow \frac{v U}{\delta^2} f''' + \frac{U^2 \delta'}{\delta} f f'' = 0$$

$$\Rightarrow f''' + \frac{U \delta'}{v} f f'' = 0$$

Note: $\delta = \left(\frac{v x}{U} \right)^{1/2} \Rightarrow \delta' = \frac{1}{2} \left(\frac{v}{U x} \right)^{1/2}$

$$\Rightarrow \delta \delta' = \left(\frac{v x}{U} \right)^{1/2} \cdot \frac{1}{2} \left(\frac{v}{U x} \right)^{1/2}$$

$$\Rightarrow \delta \delta' = \frac{1}{2} \frac{v}{U}$$

$$\Rightarrow \boxed{f''' + \frac{1}{2} f f'' = 0}$$

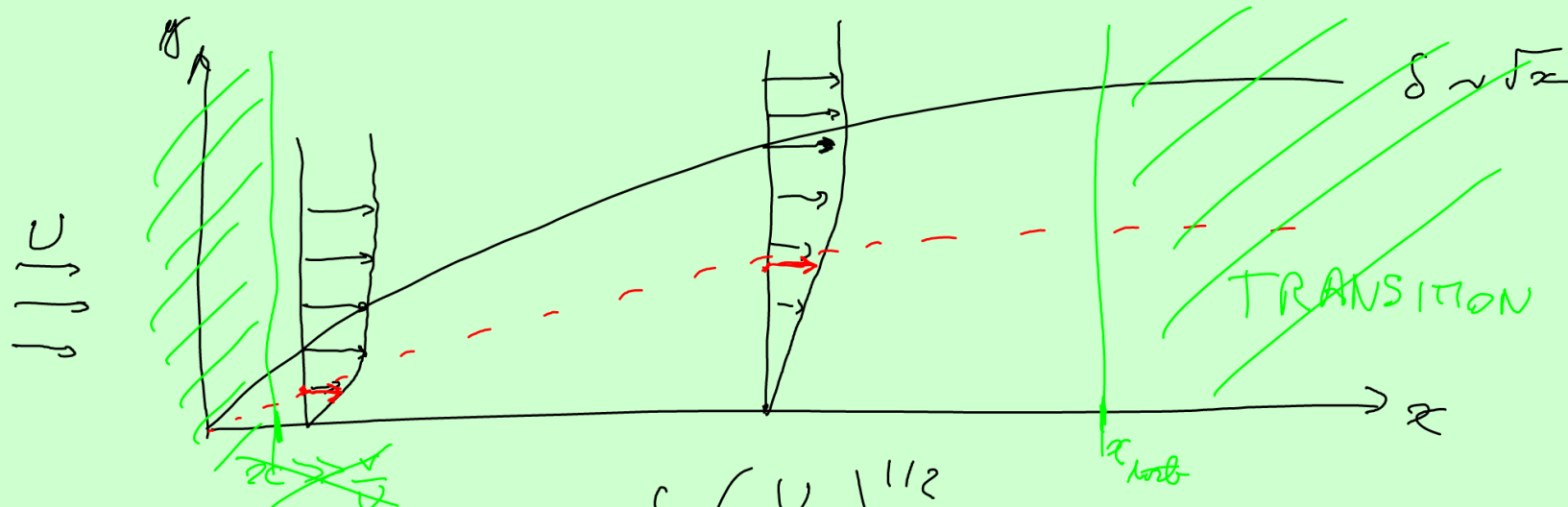
1 unknown: f
 $\rightarrow \eta$

BC: $y=0$ $u=0$
 $v=0$

$\eta=0$ $f=0, f'=0$

$y \rightarrow \infty$ $u \rightarrow U$
 $v \rightarrow 0$

$\eta \rightarrow \infty$ $f' \rightarrow 1$



$\frac{\delta}{x} = \varepsilon \ll 1$
 $\sqrt{x} \ll x$

BL below: $\eta \cong 5 \Rightarrow \delta_{99} \left(\frac{U}{\nu x} \right)^{1/2} \cong 5$

$\Rightarrow \delta_{99} \cong 5 \left(\frac{\nu x}{U} \right)^{1/2}$

$Re_x = \frac{Ux}{\nu} \gg 1$
 $x \gg \frac{\nu}{U}$

$$\Rightarrow \frac{\delta_{99}}{x} \simeq 5 \operatorname{Re}_x^{-1/2}$$

↑ aspect ratio

$$\operatorname{Re}_x = \frac{Ux}{\nu}$$

Lecture 20: from laminar to turbulence

We need the help of dynamical systems theory!

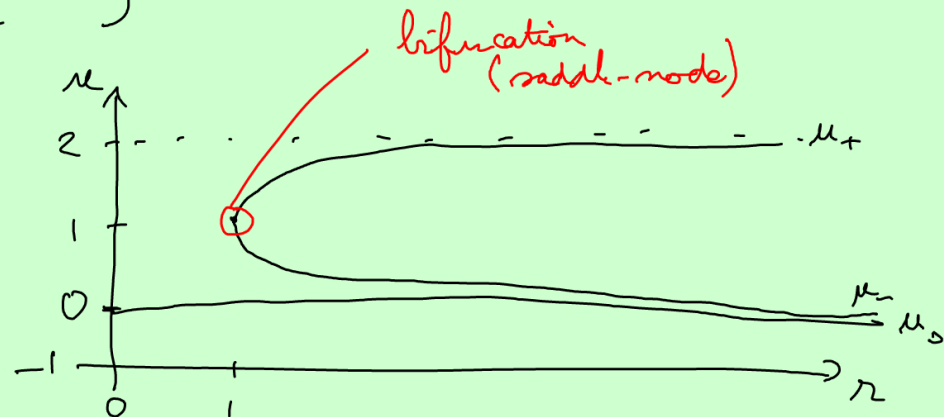
Toy problem: $\frac{du}{dx} = u \left[1 - \frac{1}{x} - (u-1)^2 \right], \quad x \geq 0$

↑ state variable
 scalar

↑ parameter

Steady states: $\begin{cases} u_0 = 0 \\ u_+ = 1 + \sqrt{1 - \frac{1}{x}} \\ u_- = 1 - \sqrt{1 - \frac{1}{x}} \end{cases} \quad x \geq 1$

bifurcation diagram:



Stability: $u = \underbrace{U}_{\substack{\text{base state} \\ (u_0, \text{ or } u_+, \text{ or } u_-)}} + \tilde{u}$ — perturbation

$$\Rightarrow \underbrace{d_x U}_{=0} + d_x \tilde{u} = (U + \tilde{u}) \left[1 - \frac{1}{x} - (U + \tilde{u} - 1)^2 \right]$$

$$\Rightarrow d_x \tilde{u} = \underbrace{U \left[1 - \frac{1}{x} - (U - 1)^2 \right]}_{=0} + U \left[-\tilde{u}^2 - 2\tilde{u}(U - 1) \right] + \tilde{u} \left[1 - \frac{1}{x} - (U + \tilde{u} - 1)^2 \right]$$

$$\Rightarrow d_x \tilde{u} = \underbrace{-U\tilde{u}^2}_{-2U^2\tilde{u}} + \underbrace{-2U^2\tilde{u}}_{+2U\tilde{u}} + \underbrace{\tilde{u}}_{-\frac{\tilde{u}}{x}} - \underbrace{\frac{\tilde{u}}{x}}_{-U^2\tilde{u}} - \underbrace{\tilde{u}^3}_{-2U\tilde{u}^2} - \underbrace{\tilde{u}}_{+2\tilde{u}^2}$$

$$\Rightarrow d_x \tilde{u} = \tilde{u} \left[-2U^2 + 2U + 1 - \frac{1}{x} - U^2 - 1 + 2U \right] + \tilde{u}^2 \left[-U - 2U + 2 \right] + \tilde{u}^3 (-1)$$

$$\Rightarrow d_x \tilde{u} = \tilde{u} \left[4U - 3U^2 - \frac{1}{x} \right] + \tilde{u}^2 [2 - 3U] - \tilde{u}^3$$

We look at small perturbations: $\tilde{u} = \varepsilon \hat{u}$, $\varepsilon \ll 1$
 $\hat{u} = O(1)$

$$\Rightarrow \varepsilon d_t \hat{u} = \varepsilon \hat{u} \left[4U - 3U^2 - \frac{1}{\varepsilon} \right] + \varepsilon^2 \hat{u}^2 \left[2 - 3U \right] - \varepsilon^3 \hat{u}^3$$

$$\Rightarrow \underbrace{d_t \hat{u} = \hat{u} \left[4U - 3U^2 - \frac{1}{\varepsilon} \right]} + O(\varepsilon)$$

We have, at leading order in ε , a first order linear ODE, so the solutions read:

$$\hat{u} = \underbrace{k}_{\text{amplitude}} e^{\underbrace{\lambda t}_{\text{temporal growth rate}}}$$

$\lambda > 0 \rightarrow$ the perturbation grows with time
 $\Rightarrow U$ is unstable

$\lambda < 0 \rightarrow$ the perturbation decays
 $\Rightarrow U$ is stable

$$\Rightarrow \lambda k e^{\lambda t} = k e^{\lambda t} \left[4U - 3U^2 - \frac{1}{\varepsilon} \right]$$

$$\Rightarrow \boxed{\lambda = 4U - 3U^2 - \frac{1}{\varepsilon}}$$

$$\text{for } U = u_0 : \lambda = -\frac{1}{\varepsilon} < 0$$

$\Rightarrow u_0$ is stable

$$\text{for } U = u_+ : \lambda = 4 \left(1 + \sqrt{1 - \frac{1}{\varepsilon}} \right) - 3 \left(1 + \sqrt{1 - \frac{1}{\varepsilon}} \right)^2 - \frac{1}{\varepsilon}$$

$$= 4 + 4\sqrt{1-\frac{1}{r}} - 3 - 6\sqrt{1-\frac{1}{r}} - 3(1-\frac{1}{r}) - \frac{1}{r}$$

$$= 1 - 2\sqrt{1-\frac{1}{r}} - 3 + \frac{3}{r} - \frac{1}{r}$$

$$= -2 + \frac{2}{r} - 2\sqrt{1-\frac{1}{r}}$$

$$= -2 \left[1 - \frac{1}{r} + \sqrt{1-\frac{1}{r}} \right]$$

$$\Rightarrow \lambda < 0 \quad \text{except at } r=1 \text{ where } \lambda=0$$

$$\Rightarrow u_+ \text{ is stable}$$

$$\text{for } U = u_- : \lambda = 4 \left(1 - \sqrt{1-\frac{1}{r}} \right) - 3 \left(1 - \sqrt{1-\frac{1}{r}} \right)^2 - \frac{1}{r}$$

$$= 4 - 4\sqrt{1-\frac{1}{r}} - 3 + 6\sqrt{1-\frac{1}{r}} - 3\left(1-\frac{1}{r}\right) - \frac{1}{r}$$

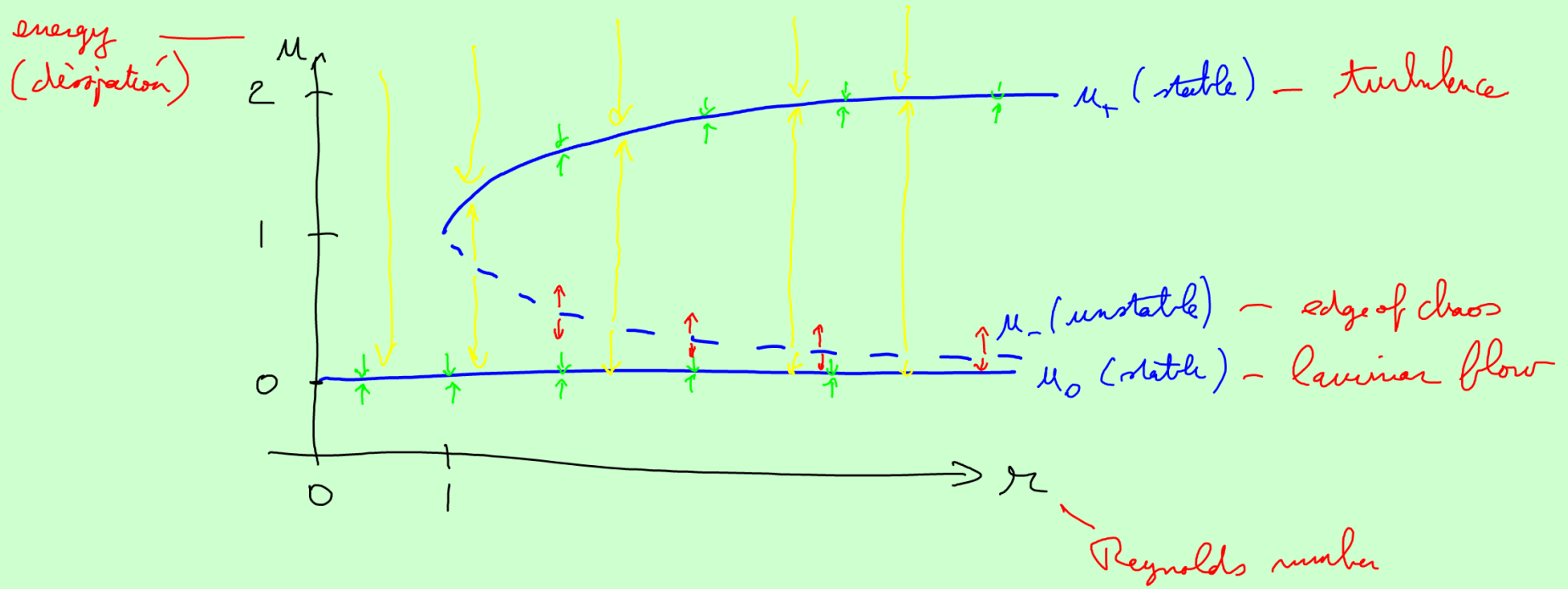
$$= 1 + 2\sqrt{1-\frac{1}{r}} - 3 + \frac{3}{r} - \frac{1}{r}$$

$$= -2 + 2\sqrt{1-\frac{1}{r}} + \frac{2}{r}$$

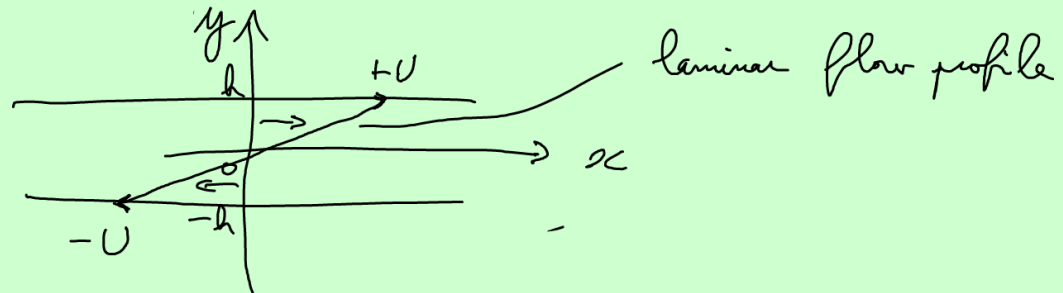
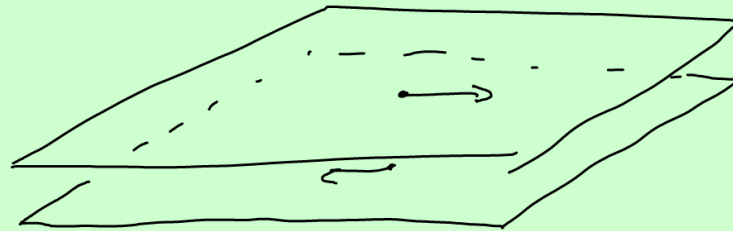
$$= -2 \left[1 - \frac{1}{r} - \sqrt{1-\frac{1}{r}} \right]$$

$$\Rightarrow \lambda > 0 \quad \text{except at } r=1 \text{ for which } \lambda=0$$

$$\Rightarrow u_- \text{ is unstable}$$



Plane Couette flow:



Navier - Stokes:

$$\frac{\partial \vec{u}}{\partial t} + (\vec{u} \cdot \nabla) \vec{u} = -\nabla p + \frac{1}{Re} \nabla^2 \vec{u}$$

$$\nabla \cdot \vec{u} = 0$$

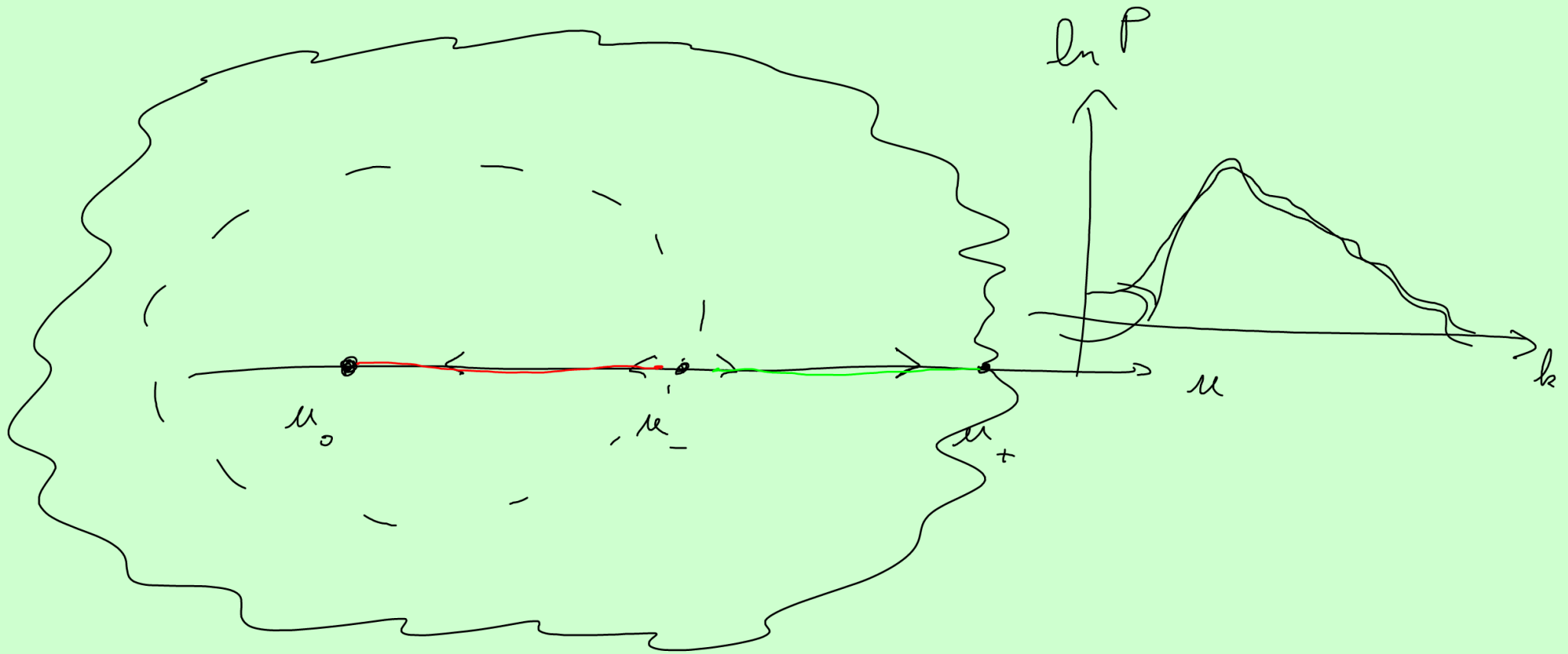
W10

BC: $y = \pm 1$, $\vec{u} = \pm \vec{e}_x$
 x periodic direction
 z periodic direction

(wall normal direction)

(streamwise direction)

(spanwise direction)



Lecture 21: High Reynolds number flows

Kolmogorov (1941)

$$\partial_t \vec{u} + (\vec{u} \cdot \nabla) \vec{u} = - \frac{1}{\rho} \nabla p + \nu \nabla^2 \vec{u}$$

|
incompressibility

look at u^2 nonlinearity

$$u \sim e^{ikx} \Rightarrow u^2 \sim e^{i2kx}$$

redistribution of the energy among
wavenumbers

brings energy to larger wavenumbers

$$\partial_t u = \nu \partial_{xx}^2 u$$

$$\partial_t \hat{u} = -\nu k^2 \hat{u}$$

$$\hat{u} = \hat{u}_0 e^{-\nu k^2 t}$$

dissipates energy
smaller structures dissipate
faster

* Large scale flow : $Re = \frac{UL}{\nu} \gg 1$

$$\text{Kinetic energy: } E_k \sim U^2$$

$$\text{Eddy turnover time: } \tau \sim \frac{L}{U}$$

Power input at large scale:

$$P = \frac{E_k}{\tau} = \frac{U^3}{L}$$

$$L^2 \cdot T^{-3}$$

* At the small scales, dissipation and inertia are in balance:

$$Re_\ell \sim 1 \Rightarrow \frac{u\ell}{\nu} \sim 1$$

$$\text{Dissipation: } \varepsilon = \nu \left(\frac{u}{\ell} \right)^2$$

$\downarrow$
 $L^2 \cdot T^{-3}$

$$\text{Using the expression of } Re_\ell: \varepsilon = \frac{\nu^3}{\ell^4}$$

$$\varepsilon = \frac{u^4}{\nu}$$

For an established flow: $P = \varepsilon$

$$(1) \quad \frac{U^3}{L} = \frac{\nu^3}{\ell^4}$$

$$\frac{U^3}{L} = \frac{u^4}{\nu}$$

$$\Rightarrow \ell^4 = L \frac{\nu^3}{U^3}$$

$$\Rightarrow u^4 = U^3 \frac{\nu}{L}$$

$$\Rightarrow \ell^4 = L^4 \frac{\nu^3}{U^3 L^3}$$

$$\Rightarrow u^4 = U^4 \frac{\nu}{UL}$$

$$\Rightarrow \boxed{\ell = L Re^{-3/4}}$$

$$\Rightarrow \boxed{u = U Re^{-1/4}}$$

Kolmogorov scale

Kolmogorov laws

* Intermediate scales, turbulence is self-similar: it will dissipate energy in proportion to the available energy at that scale.

$$\begin{array}{ccc} & \varepsilon \sim E_k & \\ / & & \backslash \\ L^2 \cdot T^{-3} & & L^2 \cdot T^{-2} \end{array}$$

NO!

$$E_k \sim v^2 \quad \text{— intermediate scale}$$

$$\varepsilon \sim v^2 \frac{v}{\pi} \quad \text{— intermediate length scale}$$

$$\Rightarrow \varepsilon \sim \frac{v^3}{\pi}$$

$$\Rightarrow v^3 \sim \varepsilon \pi$$

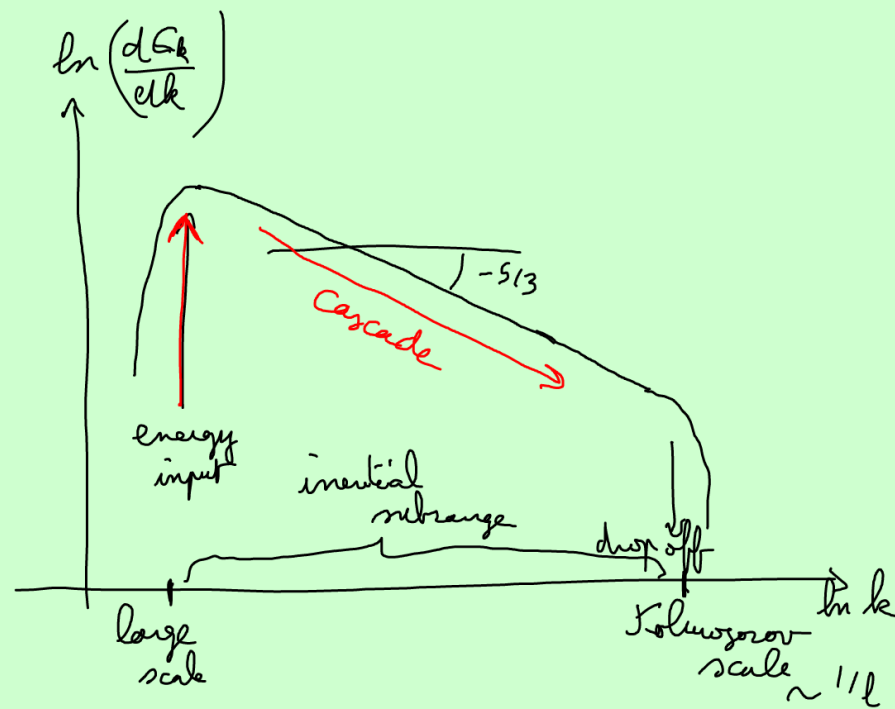
$$\Rightarrow E_k \sim \varepsilon^{2/3} \pi^{2/3}$$

$$\text{In terms of wavenumber: } E_k \sim \varepsilon^{2/3} k^{-2/3}$$

The results are generally presented using the density of kinetic energy per wavenumber:

$$\boxed{\frac{dE_k}{dk} \sim \varepsilon^{2/3} k^{-5/3}}$$

Kolmogorov $-5/3$ law



$$\partial_t u = \text{NLT} + \text{dissipation}$$

$$\frac{u^+ - u^-}{\Delta t} = \text{NLT}^- + \text{dissipation}^-$$

$$\Rightarrow u^+ = u^- + \Delta t (\text{NLT}^- + \text{dissipation}^-)$$

Lecture 22: Computational models for blood dynamics

W11

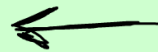
$$\left. \begin{aligned} \int_t \vec{u} + (\vec{u} \cdot \vec{\nabla}) \vec{u} &= -\vec{\nabla} p + \frac{1}{\rho} \vec{\nabla}^2 \vec{u} \\ \vec{\nabla} \cdot \vec{u} &= 0 \end{aligned} \right\} \text{DNS: direct numerical simulation}$$

RANS: Reynolds-averaged Navier-Stokes equation

$$\vec{u}(\vec{x}, t) = \underbrace{\bar{\vec{u}}(\vec{x})}_{\text{time-averaged}} + \underbrace{\vec{u}'(\vec{x}, t)}_{\text{fluctuation}}$$

$$\bar{\vec{u}}(\vec{x}) = \lim_{T \rightarrow \infty} \frac{1}{T} \int_{t_0}^{t_0+T} \vec{u}(\vec{x}, t) dt$$

$$\int_{t_0}^{t_0+T} \vec{u}'(\vec{x}, t) dt = \vec{0}$$



$$\begin{aligned} \text{Navier-Stokes:} \quad & \left\{ \begin{aligned} \rho \frac{\partial \vec{u}}{\partial t} + \rho \frac{\partial \vec{u}'}{\partial t} + \rho (\vec{u} \cdot \vec{\nabla}) \vec{u} + \rho (\vec{u} \cdot \vec{\nabla}) \vec{u}' + \rho (\vec{u}' \cdot \vec{\nabla}) \vec{u} + \rho (\vec{u}' \cdot \vec{\nabla}) \vec{u}' \\ &= -\vec{\nabla} p - \vec{\nabla} p' + \mu \vec{\nabla}^2 \vec{u} + \mu \vec{\nabla}^2 \vec{u}' \\ \vec{\nabla} \cdot \vec{u} + \vec{\nabla} \cdot \vec{u}' &= 0 \end{aligned} \right. \end{aligned}$$

average:
$$\begin{cases} \rho \frac{\partial \vec{u}}{\partial t} + \rho (\vec{u} \cdot \vec{\nabla}) \vec{u} + \underbrace{\rho (\vec{u}' \cdot \vec{\nabla}) \vec{u}'} = -\vec{\nabla} \bar{p} + \mu \vec{\nabla}^2 \vec{u} \\ \vec{\nabla} \cdot \vec{u} = 0 \end{cases}$$

$(\Rightarrow \vec{\nabla} \cdot \vec{u}' = 0)$

$$\rho u'_i \frac{\partial u'_j}{\partial x_i} = \rho \frac{\partial (u'_i u'_j)}{\partial x_i} - \underbrace{\rho u'_j \frac{\partial u'_i}{\partial x_i}}_{=0}$$

$$= \frac{\partial (\rho u'_i u'_j)}{\partial x_i}$$

$$= \vec{\nabla} \cdot (\rho \vec{u}' \vec{u}') \Big|_i$$

$$\Rightarrow \rho \frac{\partial \vec{u}}{\partial t} + \rho (\vec{u} \cdot \vec{\nabla}) \vec{u} = -\vec{\nabla} \bar{p} + \mu \vec{\nabla}^2 \vec{u} + \overline{\vec{\nabla} \cdot (-\rho \vec{u}' \vec{u}')}$$

$$\bar{\tau}^R = -\rho \overline{\vec{u}' \vec{u}'} : \text{Reynolds stress}$$

$$\Rightarrow \rho \frac{\partial \vec{u}}{\partial t} + \rho (\vec{u} \cdot \vec{\nabla}) \vec{u} = \vec{\nabla} \cdot (\bar{\tau} + \bar{\tau}^R)$$

We usually work:
$$\begin{cases} \rho \frac{\partial \vec{u}}{\partial t} + \rho (\vec{u} \cdot \vec{\nabla}) \vec{u} = -\vec{\nabla} \bar{p} + \mu \vec{\nabla}^2 \vec{u} + \underbrace{\vec{\nabla} \cdot (-\rho \overline{\vec{u}' \vec{u}'})} \end{cases}$$

$$\vec{\nabla} \cdot \vec{u} = 0$$

$$\text{Residual: } \begin{cases} \rho \frac{\partial \vec{u}'}{\partial t} + \rho (\vec{u} \cdot \vec{\nabla}) \vec{u}' + \rho (\vec{u}' \cdot \vec{\nabla}) \vec{u} + \rho (\vec{u}' \cdot \vec{\nabla}) \vec{u}' - \rho \overline{(\vec{u}' \cdot \vec{\nabla}) \vec{u}'} = -\vec{\nabla} p' \\ \vec{\nabla} \cdot \vec{u}' = 0 \end{cases}$$

$$\text{Indices: } \rho \frac{\partial u'_i}{\partial t} + \rho \bar{u}_k \frac{\partial u'_i}{\partial x_k} + \rho u'_k \frac{\partial \bar{u}_i}{\partial x_k} + \rho u'_k \frac{\partial u'_i}{\partial x_k} - \rho \overline{u'_k \frac{\partial u'_i}{\partial x_k}} = -\frac{\partial p'}{\partial x_i} + \mu \nabla_k^2 u'_i$$

$$\text{Multiply by } u'_j: \rho u'_j \frac{\partial u'_i}{\partial t} + \rho \bar{u}_k u'_j \frac{\partial u'_i}{\partial x_k} + \rho u'_k u'_j \frac{\partial \bar{u}_i}{\partial x_k} + \rho u'_k u'_j \frac{\partial u'_i}{\partial x_k} - \rho u'_j \overline{u'_k \frac{\partial u'_i}{\partial x_k}} \\ = -u'_j \frac{\partial p'}{\partial x_i} + \mu u'_j \nabla_k^2 u'_i$$

$$\text{Swap } j \text{ and } i: \rho u'_i \frac{\partial u'_j}{\partial t} + \rho \bar{u}_k u'_i \frac{\partial u'_j}{\partial x_k} + \rho u'_k u'_i \frac{\partial \bar{u}_j}{\partial x_k} + \rho u'_k u'_i \frac{\partial u'_j}{\partial x_k} - \rho u'_i \overline{u'_k \frac{\partial u'_j}{\partial x_k}} = \dots$$

$$\text{Sum: } \rho \frac{\partial (u'_i u'_j)}{\partial t} + \rho \bar{u}_k \left(u'_j \frac{\partial u'_i}{\partial x_k} + u'_i \frac{\partial u'_j}{\partial x_k} \right) + \frac{\partial \bar{u}_i}{\partial x_k} \rho u'_k u'_j + \frac{\partial \bar{u}_j}{\partial x_k} \rho u'_k u'_i \\ + \rho u'_k \left(u'_j \frac{\partial u'_i}{\partial x_k} + u'_i \frac{\partial u'_j}{\partial x_k} \right) - \dots$$

$$\begin{aligned}
=> \frac{\partial(-\tau_{ij}^R)}{\partial t} + \underbrace{\rho \bar{\mu}_k \left(\mu_j' \frac{\partial \mu_i'}{\partial x_k} + \mu_i' \frac{\partial \mu_j'}{\partial x_k} \right)}_{\rho \frac{\partial (\bar{\mu}_k \mu_i' \mu_j')}{\partial x_k}} + \frac{\partial \bar{\mu}_i}{\partial x_k} (-\tau_{kj}^R) + \frac{\partial \bar{\mu}_j}{\partial x_k} (-\tau_{ki}^R) \\
& + \underbrace{\rho \mu_k' \frac{\partial (\mu_i' \mu_j')}{\partial x_k}}_{\rho \frac{\partial (\mu_i' \mu_j' \mu_k')}{\partial x_k}} + \dots = 0 \\
& \underbrace{\rho \frac{\partial (\mu_i' \mu_j' \mu_k')}{\partial x_k}}_{\text{3rd order moment}} - \cancel{\rho \mu_i' \mu_j' \frac{\partial \mu_k'}{\partial x_k}}
\end{aligned}$$

3rd order moment
 $\rightarrow$ need a closure!